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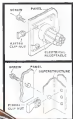
July 15, 1963

SPECIAL REPORT:

**Gemini
Spacecraft
Computer**

USAF Thunderbirds
Crossover at Paris





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P15004



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AEROSPACE CALENDAR

(Continued from page 1)

- American Institute of Aeronautics and Astronautics, Houston, TX
- Aug. 19-21—Aerodynamics Conference, American Institute of Aeronautics and Astronautics, Yale University, New Haven
- Aug. 19-21-1961—Computer Engineering Conference, Boulder, Colo., Sponsored by University of Colorado, NBS Computer Engineering Laboratory
- Aug. 20-23-1961—Western Electronic Show and Convention (WEECON), Civic Auditorium, San Francisco, Calif.
- Aug. 20-25—Simulation for Aerospace Flight Conference, American Institute of Aeronautics and Astronautics, Delta-Hilton Hotel, Columbus, Ohio
- Aug. 20-25—Conference on Physics of Earth, Oak Ridge, Tennessee, American Institute of Aeronautics and Astronautics, Massachusetts Institute of Technology, Cambridge, Mass.
- Sept. 4-11—International Symposium on High-Temperature Technology, Adelaide, Calif., Sponsored: Stanford Research Institute
- Sept. 8-11—Annual Meeting, Air Industries Association of Canada, Montreal, Quebec, Que. Canada
- Sept. 9-11—Seventh National Conference on Military Electronics Institute of Electrical and Electronics Engineers, Sheraton Hotel, Washington, D.C.
- Sept. 9-12-1961—Annual International Aerospace Conference & Exhibit, Technical Society of America, McCormack Place, Chicago, Ill.
- Sept. 10-12—National Symposium on Space Radiation, Science and Recovery, Edwards AFB, Calif., Sponsored: Aerospace Environmental Society, Air Force Flight Test Center
- Sept. 10-12—New York University's Third Annual Air Transport Conference, Waldorf Astor, New York, N.Y.
- Sept. 11-15-1961—Annual National Civil Aviation & Aerospace Symposium, Air Force Association, Sheraton Park, and Sheraton Hotel, Washington, D.C.
- Sept. 16-18—International Aviation Research and Development Symposium, Atlantic City, N.J., Sponsored: Federal Aviation Agency
- Sept. 18-19-1961—Aircraft Operations and Maintenance Symposium, Midway, N.Y.
- Sept. 19-18—North Atlantic Treaty Organization Trade & Travel Fair, San Francisco, California, Sheraton Hotel
- Sept. 21-27—National Aeronautics and Space Engineering and Manufacturing Meeting and Display, Sheraton Hotel, Los Angeles
- Sept. 27-27—International Telecommunications Conference, Sheraton Hotel, London, England, Sponsored: Institution of Electrical Engineers (London), American Institute of Aeronautics and Astronautics, Institute of Electrical and Electronics Engineers, Technical Society of America
- Sept. 27-28—Second Annual Symposium on the Physics of Matter in Electronics, Chicago, Ill., Sponsored: Atomic Energy Development Center, Atomic Research Foundation
- Sept. 28-Oct. 1-1961—Congress, International Astronautical Federation, Paris (Continued on page 9)

Pulse magnetron, used in commercial air weather radar systems, are part of the extensive line of Litton microwave tubes and display devices. For information write to San Carlos, California. In Europe, Box 110, Zurich 50, Switzerland

LITTON INDUSTRIES
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PROBLEMATICAL RECREATIONS 179



Furbacher lives in Chicago Acres and works in Beverly Hills. LaRonde lives in Beverly Hills and works in Chicago Acres. They usually leave their respective homes at the same time, and pass each other at Sun's Hamburger Shack. (Furbacher drives faster to Sun in LaRonde.) On a day when Furbacher's wife Frances spoiled the eggs and he left 5 minutes late, they passed each other at the gas station, two miles from Sun's. How fast do they drive?

—Cassidy
Announcing a new model Decline (lighter and more compact) for each service environment as mobile, aircraft, and truck-mounted installations. This new alpha-numeric readout device used to monitor or display digital data weighs only 5 ounces and has been reduced in depth by 3 to 4 1/2 inches. Costs less, too. Get the full scoop from our Product Marketing Department, Guidance and Control Systems Division, 2590 Canoga Ave., Woodland Hills, Calif.

ANSWER TO LAST WEEK'S PROBLEM: R = 18, Area = 1296, Volume = 7776

LITTON INDUSTRIES, INC.
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defeated for high speed search, direct recording to 300 KC phase and amplitude equalized, FM recording to 20 KC phase equalized, and all solid state electronics throughout. There's nothing around that will measure up to the FR-1300 for performance per pound. And one more thing: Ampex precision and reliability through and through. For data write: Ampex Corp., Redwood City, Calif. Worldwide sales, service.



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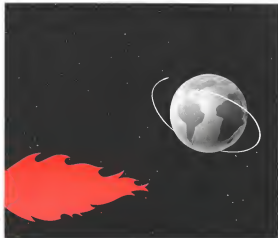
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EDITORIAL

More Supersonics

The competitive pressures in the supersonic transport field are beginning to build up to explosive levels. We predict that by the time the 19th annual general meeting of the International Air Transport Association convenes next October in Rome, there will be ample cause for some of the delegates to lamentate over the "rush-jump frenzy" in the supersonic equipment field, just as they have been doing for the past few years over the subsonic jet picture. We think these warnings will have about the same effect on supersonics as they had on the initial sound of subsonic equipment buying—nil.

The airlines operating in sharply competitive markets where only a sharp edge of efficiency can keep them in the black, will certainly seize on the first proven supersonic jet transports, whether they are in the Mach 2 or Mach 3 speed range. Juan Trippe's recent admission (AW June 18, p. 48) that Pan American World Airways will take the plunge is already having significant effects among his competitors. Most of them will have to follow his lead within the next few months, whether they really want to jump into supersonics or not, simply to maintain their competitive position and avoid being dubbed as "obsolescing" airlines.

Concorde Lead

Meanwhile, it is worth reviewing the international competition in the race to capture this market to determine if there are any significant trends among the competitors that may, hereinafter, be newer. The Anglo French Concorde Mach 2.2 project is off to a good start, has avoided many of the standard European pitfalls and must be considered the current leader. The British and French technicians face many problems in bringing their aircraft to operational status. This ever flies are working well within the current state of the accumulated art where the French, particularly, are gaining considerable Mach 2 delta flight experience with their Mirage series aircraft. We expect their biggest problem will be simply the slower European pace and the lack of sufficient research and development facilities on the scale required for an effort of this magnitude. Europeans tend to under estimate U. S. speed in its development pace once technical goals have been fixed, set and managerial and governmental support is determined. They should watch the current race between the de Havilland Trident and the Boeing 727 for an example of what we mean by this.

The Soviet Union plans to compete in the supersonic transport era, but we tend to discount the significance of its effort in any international competitive arena.

The Soviet's jump into the jet transport age has been a remarkable achievement for their technicians and industry. But they have not been able to sustain much status when stacked against Western competitors. None of the Soviet jet transports has proved sufficiently economical in airline operation to be attractive even to other countries with Communist governments. The Chinese, Poles and Yugoslavs have all purchased transports of Western manufacture when possible. Even the African nations who exchanged their peasant crops for Russian and American transports have found the Soviet aircraft uneconomical in competitive markets.

Soviet Handicap

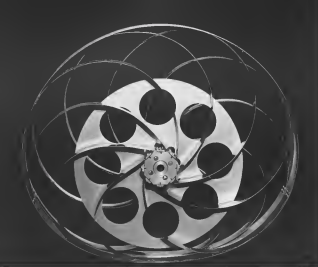
Much of the Soviet defenses in this regard can be credited to the rigid security measures in which their technicians worked for the past 30 years. This left them rather naive in those areas where outside sales competitive contact with the West would have educated them thoroughly. According to the latest intelligence coming in from Moscow, Semyonov is heading the Soviet supersonic transport program and has settled on a large delta-wing design with four large turbojets mounted in two nacelles under the trailing edge of the wing. His general plan is similar to the Anglo French Concorde design. This should come as no surprise to those who have noted prior evidence of Mr. Semyonov's regard for British design in the striking resemblance between his 1962 and the British Aircraft Corp. VC 10. The Soviet design is also aimed at the Mach 2 speed range.

We doubt that the Soviet supersonic transport will have much competitive impact outside the ports of Poland and Leningrad. But we are also certain that it will eventually appear on the Soviet international routes in the form of Aeroflot as the very best effort the Russian aircraft industry can produce.

The biggest mystery in this three cornered competition is still the United States. If this country does some firm goals within the next few months and gives its technicians a clear task without too many political obstacles, we think the other competitors will be surprised at the pace with which the U. S. development program will move, and the European lead may evaporate fast.

However, it is possible to minimize the debate, reality and political log-rolling just the point where even the finest technical resources in the world can ever regain the critical lost time. This is the knowledge which the U. S. supersonic transport prospects are now balancing.

—Robert Blatz



WE JUST INVENTED THE WHEEL, AGAIN . . . Wheels are the most popular locomotion mechanism on earth today. How about on the moon? Considering the harsh lunar environment, the unknown characteristics of its terrain and the questionable soil strength, are wheels also suitable for locomotion on its surface? Bendix engineers are seeking the answer. As part of the company sponsored Lunar Roving Vehicle Program, Bendix has developed several lunar wheel configurations with such desired characteristics as reliability, efficient traction on a wide range of soil conditions, quick damping reaction and negotiability over and around obstacles—all at minimum weight. If you are an engineer or scientist in the space technologies, and would like to join this team for greater personal recognition and opportunity, please contact our Personnel Director, Bendix Systems Division, Ann Arbor, Michigan. An equal opportunity employer.

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Washington Roundup

New Briefing Control

Defense Dept. within the next few weeks will issue new rules designed to tighten briefing control over leading military services and industry on future projects.

Portugal spokesmen declare they are merely trying to bring order into the chaotic briefing procedures of the Air Force, Navy and Army. But more military leaders see the rules as a further tightening of their check-rein to Defense Secretary Robert S. McNamara.

Staff of the Defense Industry Advisory Council initiated the move with the approval of Deputy Defense Secretary Russell L. Gilpin. Now Dr. Harold Blom, director of defense research and engineering, has the job of implementing the new briefing format. He is expected to issue rules in the coming weeks, the end of August, perhaps as early as this month.

Leaders of the new rules complain that the services in the past squandered huge military budgets by conducting elaborate contract or concept proposals which had no chance of being approved at the top. This, they argue, wasted industry's time, and money on studies and proposals. Also, the industrial services have not coordinated their briefings, even with or with, their own administrative staffs.

The new rules will require the services to get approval from DOD's before asking industry to discuss their future needs. As more streamlined, the rules on briefings will apply only to exploration, projects—not specific items, near the procurement stage. Defense Dept. also hopes the rules will ensure that the services brief the pertinent segment of industry, not just a few favored firms. Representatives of the defense services are expected to attend the briefings under the new format. "We've got to get a handle on what they're doing," one defense official said in preparing the rules.

Webb's New Approach

Blazing the House space committee for sharply cutting his budget marks a radical change in tactics for Administrator James E. Webb of the National Aeronautics and Space Administration. (ENR 3/31)

Webb's leadership has pushed himself on working his will on Congress through steady but quiet consultation with the leaders. Last week he started to frontal assault, changing the committee's authorization bill—which sets ceilings for appropriations—amounted to an unprecedented level of support.

But the biggest battle is yet ahead when legislation actually appropriating the money to NASA is voted. Chairman Clarence Cannon of the House Appropriations Committee has no little criticism for the space program that he again will meet on deep cuts in the budget. Last year he considered cutting the NASA budget \$145 million, a slash \$77 million greater than that recommended in the House authorization bill.

New JFK Helicopters

Despite some congressional opposition, President Kennedy soon will have two additional Sikorsky VH-3A helicopters for the White House fleet. The additions will bring the fleet total to 10 VH-3As, five from the Army, and five from the Marine Corps. The new aircraft will replace two VH-3As. The VH-3As will cost \$1.125 million each and be paid for through appropriations of Army and Navy programs in fiscal 1965. Opposition in Congress, which now appears large, revolves around the question of whether the two VH-3As were really needed. President Kennedy wrote a personal letter requesting the helicopters. Part of the justification was that the helicopters might be needed in an emergency.

Korth's Woes Mount

Navy Secretary Fred Korth is swimming in a sea of troubles these days. His military leaders resent the part he played in the firing of Adm. George Anderson as chief of naval operations. In testimony before the Senate subcommittee, mentioning the F-111 (ENR) contract and rival Chairman John L. McClellan, he said he is for a continuing, painful benefit examination to see if his activities as secretary suited an official selected role.

Korth was president of a boat in Ft. Worth, Tex., that had dealings with General Dynamics, winner of the F-111 contract. He told the Senate subcommittee he stayed objective and did not take a leading role in the F-111 negotiations to avoid any appearance of favoring his former business connection. But after saying that, Korth told the subcommittee in great detail the technical issues which figured in his preference for General Dynamics.

Another conflict-of-interest problem that some senators want to explore is what part Korth played in the award of the X-22A contract to Bell after the Navy preference, but recommended Douglas. Korth had been a director of Bell. Gilpin's remarks told a Senate subcommittee that he asked Korth about Bell's management before awarding the X-22A contract (AW June 24 p. 38).

The reason that NASA's launch of an Air Force payload with a Scout booster from Wallops Island, Va., June 28 was kept as secret was that the USAF and space agency press officers specially assigned to the detail said thought the other was informing the press.

—Washington Staff

GE Lab Uses Rodent To Power Transmitter

General Electric's Space Sciences Laboratory recently operated a 500-kg transmitter continuously for eight hours using electrical current generated in the body of a living rat in the role source of power.

Power was developed across an electrode implanted under the rat's skin and a second one located in its parietal cavity. No biological functions were disturbed in the process, which resulted in a maximum power of 145 microamps with a 500 ohm load at 0.23 v. Open circuit voltage rose to 0.85 v.

The experiment, financed by a recently awarded \$77,538 contract from National Aeronautics and Space Administration's Aeronautics Research Center, is part of a two-year study of bioelectronic which grew out of a biological fuel cell development program at GE. One biological fuel cell developed prod-

uces electric power from the activation of food and glucose.

Under the NASA contract, GE will attempt to develop power sources for aircraft and instrumentation that could be used with experimental animals. GE will observe power production from rats and larger animals over test periods of at least six months, in an effort to determine if detrimental physiological effects will result and if electrical current being generated remains constant.

In one experiment under air load conditions, electrodes were left in a rat's body for six months—across the average lifetime of a healthy animal—with no harmful effects in the animal. Voltage across the electrodes remained the same at the time the electrodes were implanted.

The generation of electricity in living organisms—called bioelectrogenesis—is all current in medical and biological researchers, since recent advanced radioelectronics devices, using a body's

V-107 Leasing

New York—Pan American World Airways has agreed to buy two Boeing V-107 twin-engine helicopter and lease them to New York Airways, the city's helicopter airline (AW May 11, p. 30).

Included in the agreement is a provision for helicopter service from the roof of the new Pan Am Building in midtown Manhattan to the Pan Am tower at Idlewild Airport. Approval of the leasing service is still pending before the Federal Aviation Agency and city authorities (AW May 27, p. 36).

bioelectric potentials (AW May 27, p. 36) allow electronic devices implanted in the body to regulate its functions or to stimulate physiological reactions without using an externally connected wire.

Increasing Importance of Space Acknowledged in New DOD Post

Washington—Increasing importance of the role played by space in defense

missions was recognized last week in the creation of the post of deputy director of defense research and engineering for space technology. Dr. Albert G. Yell, 49, vice president and general manager of the Minutia Co.'s Space Systems Div., was named to the post.

The post formerly was that of special assistant to the director, Dr. Harold Brown. The incumbent, Dr. Lawrence L. Korman, will leave at the end of the month (AW Apr. 15, p. 71).

The Air Force, which is charged with responsibility for most of the principal military space missions, long has recognized that technical and operational control of space technology at the defense operational level was splintered and

lacked the depth of a title of sufficient stature.

Dr. Hall, as a deputy, will have policy and monitoring responsibilities over space exploration development, advanced development and engineering development along with responsibility for acquisition and support of weapons and world wide tracking network.

Cognizance over major space operational matters, however, will rest with Dr. Brown and Dr. Eugene C. Fisher, who in the No. 2 position is principal secretary of defense—deputy director. He is formerly deputy director for research and information systems and in that position had similar space responsibilities.

Current assignments within Dr. Brown's organization, which are being

studied to match job descriptions with personnel abilities and interests, will soon be completed. The variety of scientific and engineering personnel, which grew at two-year intervals, has now scaled.

The number of dignities will increase as the business of the organization. The deputy director for engineering and chemistry were reestablished after the departure of the current deputy, Dr. James H. Goodson.

Dr. Hall was graduated from Texas A&M in 1916 with a degree in electrical engineering and received his master's and doctor's degrees from the Massachusetts Institute of Technology in 1920 and 1924. He was an early radio and wire engineer for research guidance while at MIT from 1929 to 1930. He joined the Borden Corp. as associate director of the Research Laboratories Div. in 1931, and became general manager of the scientific division of the division in 1954.

After joining Martin in 1948 as director of engineering at the Defense Div., he supervised the technical development of the Titan missile. In 1960 he became Martin's vice president managing at Baltimore, and was named to his present position in January, 1962.

Dr. Hall will have five assistants: one in Eng. Gen. Paul F. Cooper, USAF, assistant director for major and space ground support. Title of the chief assistant physicist has not been announced but the post—workful of present—will cover the technology and development aspects of space.

USAF Establishes Gemini Liaison Office

Washington—Air Force, last week announced it is establishing a three-man Gemini liaison and support office at the U. S. space agency's Manned Spacecraft Center in Houston (AW Dec. 10, 1962, p. 13 Jan. 25, p. 20).

Air Force had hoped to have a 200-man Gemini group at the National Aeronautics and Space Administration's Houston facility. In fact, and a partial joining with the Gemini flight program.

So far, no agreement has been reached between the Air Force and NASA on Air Force participation in the program. Defense Dept. officials have indicated that Gemini development is now too far along for modification necessary to experiment with manned satellite inspection, intercept and surveillance systems.

The USAF Gemini office being established at Houston under Col. William A. Hoffmann will coordinate mail pouches with the Titan 2 launch vehicle for Gemini and recover capsules.

Draken Modified for Reconnaissance Mission

New photo reconnaissance version of Sweden's Saab 370 Draken Mach 2 interceptor has a modified camera carrying nose in place of the interceptor's radome. The modified aircraft, designated S370L, shown on this and the opposite page is scheduled to replace Saab S370C reconnaissance aircraft presently in service with the Royal Swedish Air Force. The S370L is equipped for both high and low-level missions. Close-up photos at the nose section, opposite page, show the location of various camera ports all of which are fitted with defocusing devices. Note the apparently matched windows on

the windshield, probably used for stereoscopic or wide-angle photography. The status nose slides forward to present rapid access to the camera and ray correction of exposed film magazines after the mission. Nose operation is located at the rear port behind the side port, evident in the close-up photo. Such view that a new Swedish camera sight and multivision navigational equipment have been incorporated into the S370L. The new Draken version made its first flight last month. Saab Test Pilot Center Ulfareden was at the controls on the maiden flight.



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Japan Begins Missile Development Tests

Tokyo—Japanese defense agency last week successfully launched two air-to-air missiles and one surface-to-air missile from Niihara Island onto the Pacific. The shots were part of an 11-launch test series scheduled to last until Aug. 5.

Five different types of missiles, all manufactured by Mitsubishi Heavy Industries, Rooman, 214, are scheduled in the series. Test launches will be from ground installations.

The two air-to-air missiles launched were TAAM-2-0s (TAM-2 designation test item). The TAAM-2-0 missile is about 9.1 ft. long, 5.9 in. in diameter and weighs about 141.5 lb. It has four trapezoidal guidance fins mounted on the rear, forming an easily symmetrical cross.

The solid-fueled TAAM-2-0, which developed 4,400 lb thrust for 1 sec., was launched on ballistic trajectory at 45 deg angle. Minimum altitude at launch by the missile was 6,560 ft. Downrange distance attained was slightly under 4 mi.

The surface-to-air missile launched was a two-stage TLRM-2D. The TLRM-2D is 39.2 ft. long, 16.5 in. in diameter and weighs 1,065 lb. Its booster stage has a double base solid fuel which produces 12,540 lb thrust for 1.5 sec. Section stage was a dummy for the test shot. Both stages use control fins to respond to the booster and deballasted on the ramjet-actuated air-to-air reentry vehicle.

The TLRM-2D (D designation) was launched on a ballistic trajectory at a 50 deg angle. The booster and stage separation function. The booster stage reached an altitude of 15,500 ft and limited about 2 sec downrange. The reentry stage landed near a mile and a half from the test area.

Four XAAM-A-5 missiles, improved infrared homing versions of the TAAM-2-0 missile, also will be launched. The XAAM-A-5 (X designation experimental) is 39.2 ft. long, 16.5 in. in diameter, weighs 1,065 lb. It has a double base solid fuel which produces 12,540 lb thrust for 1.5 sec. The XAAM-A-5 has four trapezoidal guidance fins set in the rear and four deballasted guidance fins on the front. The aftward-homing system controls the front fins through a gas-sensit linkage. Propellant is a solid double base fuel which produces 6,092 lb of thrust for 2.5 sec.

The four test missiles will be launched at 15 deg on ballistic trajectory downrange designed to carry them about 4 mi downrange.

The reentry four launchers will be with two TLRM-2, boost using, one-

face-to-air missiles and two TSAM-1 surface-to-air missiles.

The TLRM-2D boost ride is identical to the TLRM-2D, but the reentry stage will be fueled with liquid nitric acid and acetone. Section stage deballasted on the ramjet-actuated air-to-air reentry vehicle. The missile follows a radio beam, with hydraulic nozzle actuating the rear fins for directional control. Planned launch angle for the shot, which will test all systems, is 50 to 54 deg.

The TSAM-1 surface-to-air missile has preprogrammed reentry guidance. This also is a two-stage missile and has the same dimensions as the TLRM-2D, but weighs slightly less. The booster

has trapezoidal stabilization fins and the reentry has deballasted fins front and rear.

Propellant in the booster stage is a double base solid fuel which develops 12,540 lb of thrust for about 3.5 sec. The reentry stage has a double base solid fuel engine burning liquid nitric acid and acetone. It will develop 7,700 lb of thrust for about 10 sec. Guidance control on the reentry is achieved with hydraulic nozzle which actuates the rear fins.

Planned launch angle for the TSAM-1 shot is 50 to 70 deg and downrange flight is expected to be about 12.5 mi. Launchers will include full vehicle shockloads.

French Conduct Re-entry Tests

By Cecil Brownlow

Paris—Mach 12 re-entry tests to evaluate the proposed new core configuration for France's independent atmospheric range ballistic missile (IRBM) weapon system have been started. Launches will be from the test area's main to develop a reentry vehicle, subsonic-boosted IRBM by the late 1960s.

France's Office National d'Etudes et de Recherches Aérospatiales (ONERA), which works closely with the Ministry of Defense, reported last week that two such flights have taken place thus far using the test stage. Reentry research vehicle is a booster. Results of the tests were described in very satisfactory.

New core used in the flights from the new test launch facility on the Des de Levant island chain near the Mediterranean coast of Tunisia was designed and built by Sud Aviation. The solid-propellant booster was developed by ONERA in collaboration with the Société d'Etude de la Propulsion par Réacteur à Solid (SERP).

First Mach 12 test was made on June 26 with a primary objective of measuring the heat flux along the nose cone during its reentry into the atmosphere. Second launching on July 3 was designed to evaluate the core's protective covering under the time loss conditions at those measured during the initial flight.

Total of eight Mach 8 tests of the new core will be conducted presently with the ONERA Airborne Mach 8 reentry platform of the Berrin.

First two stages of Berrin for the core to an altitude of approximately 170 mi. Last two stages join the pre-

liminary in a maneuver simulating those performed by the Lockheed X-17 booster vehicle during the U.S. reentry tests in the late 1950s—once allowed to an even and fall back to an altitude of about 94 mi before the last two stages fire downwards in rapid succession to propel the core into a Mach 12 speed for the reentry simulation.

Guidance, a major task in view of the proximity of the maneuver to the test area, is maintained during five-stage sequence by means of a three-axis autopilot controlling four servos reentry vehicle attached to the base of the nose. Once the vehicle is on course and after first-stage burnout, stabilization is maintained during the remainder of the reentry sequence by four starman fins attached to the second stage. Third and fourth stage stabilization is aided by the control system housing the reentry simulation points.

Downrange traveled during the 5.5 min of flight is approximately 70 mi. An optical tracking system linked to an electronic distance command radar measures the flight during the initial 30 sec and on continuously until the rocket is destroyed at 100 mi or less. No air intercept has been made to recover the two nose cones after their reentry, but onboard equipment maintains the core's position and transmits the data to an ONERA ground station.

Berrin itself has a launch weight of 7,400 lb, substantially lighter than its 15,000 lb Airbus predecessor. Overall length is 43 ft.

With some modifications, the Berrin could be used as a sounding rocket in France's national space research program. The rocket, which has four stages, guided during ascent could reach a 110 ft per second to an altitude of 630 mi.

Atlas Fluorine-Oxygen Oxidizer Studied

National Aeronautics and Space Administration is considering a program to determine the feasibility of using a mixture of fluorine and oxygen (Fluo) as an oxidizer in the USAF General Dynamics/Aerometrics Atlas space launch vehicle (SLV).

Fluo is a mixture of fluorine and oxygen. Possible benefits include:

• Varying possible on-orbit use which the Atlas has already been altered as the booster.

• Use of the Atlas as a possible backup for the Martin T-2 booster in the Gemini program.

NASA was briefed June 30 by GDA personnel on the Atlas potential with use of the fluorine-oxygen combination as an oxidizer.

Propulsion of approximately 79% fluorine to 50% oxygen by weight is the optimum ratio for specific impulse gain with the P-4 fuel used with the Atlas.

Estimates are that at 10-15% fluorine in specific impulse gain is realized with this proportion.

However, use of 70% fluorine by weight probably would require such redesign as Atlas' structural changes that the results might be considered as often a new engine. Lesser percentages of fluorine, down to 50% or less than 10%, also might require extensive changes in the engine.

A combination of 10% fluorine by weight probably would require only minor modifications to the Atlas launch vehicle. This amount of fluorine in the oxidizer would result in a specific impulse gain of 3,500 mph, providing a booster capacity for an additional 12,000 lb payload to a 100-mi orbit. (Atlas' current 10% Fluo) fluorine in the oxidizer would be required to achieve an orbit with Gemini.

If this concept is implemented, first launch in the fluorine program could be accomplished in a year and would require considerable production, testing with the reentry configuration. Vandenberg AFB, Calif., might be the launch site.

News Digest

State Dept. last week expressed hope that Britain, Spain, Greece and Mexico would further isolate Cuba by denying transit rights to transport aircraft operating to and from the Communist island. Spain's Iberia Airlines is operating direct services to Havana. Cuba and Czechoslovakia remains an operating throughway. Canada, Mexico and Bolivia's Gran Canaria Island in the Caribbean U.S. is concerned that Cuban Communists are entering South America via Grand Canaria.

NASA Joseph A. Walker, chief NASA test pilot, took the No. 1 shuttle to an altitude of 223,000 ft, reaching a maximum speed of 3,800 mph after an 84-sec. engine burn time. The retractable 14-shaped probe, mounted near the tail of the X-15, extends 3 ft. This was the third test flight with the instrument, which is being developed by the team of the Air Force and the Navy to monitor the surface of the ship.

First British Atlantic Cap. 3-4C 111 two-seat transport starts engine runway tests this week. Royal Navy's first flight will be made during the last week in July or the first week of August.

Eleventh Minuteman ICBM was launched successfully recently from a silo at Vandenberg AFB. The flight, a research and development flight, operated purely by Boeing and USAF's Ballistic Systems Div. The flight was conducted to test overall missile operation, provide sub-orbital data and valid data representation in various components.

Minuteman has been awarded a second one-year contract valued at \$11.7 million for support services at the Marshall Operations Plant in New Orleans.

As few measurements with a traversing probe were conducted during a July 9 flight of the NASA/NASA-Aerometrics

will consider results of three to 16 air-to-air shots during the mission, including a Man-at-Arm of 10 to 10 days.

Mitsubishi Space Flight Center last week awarded a \$10,000 contract to Space Technology Laboratories, Inc. for a two-stage pump study of mobile stage concepts for future space operations.

Soviet Union has conducted test of space launch vehicles in the Central Pacific. Test last week. The second test in the Central Pacific is expected by the USSR for launch impact area (AW) May 20. A 100 has been reported for ships and aircraft. Test failed.

Andrew S. Hershman has been named corporate controller for Aero-Venture Corp. He previously was the company's assistant controller.

Wig-Gun, Frank P. Baker, the second official Army armor chief last week at the age of 55. William Wright taught Gen. Cohen to fly in 1909 as an early Wright biplane, the first aircraft produced by Army.

Dr. Nello Paner last week was appointed a consultant to National Aeronautics and Space Administration to study and analyze the reentry program. He will report directly to NASA Administrator James E. Webb. Dr. Paner is professor of physiology at the University of California.

United Aircraft Corp. is offering sub-orbital rights to conduct studies for \$42,504,000 in non-refundable debentures. Funds raised in the offering, underwritten by Hercules Supply & Co., will be used to retire short-term bank loans. United has forecast that its cash requirements this year would require additional short-term borrowing, but the firm is to start term bank debt from \$4.5 million at the beginning of 1962 to \$14 million at the end of the year. United also \$15 million in June 30, against a loss of credit of \$70 million, apparently caused the debt to long term financing.

Genital Sulf Tests

R. F. Goodrich Co. will begin rifle M16 tests next month on the Green's open rifle. Four Green's rifle variants will be tested by the David Clark Co., Inc., will be used in the test program to determine how well modern weapons perform under adverse conditions, including water and glove operations perform in repeated jumps simulating 10 minutes of 10 days only.

Technicians Score FAA on Traffic Control

RTCA report is highly critical of agency for failure to maintain pace with growth of air traffic volume.

Washington—Federal Aviation Agency's management of air traffic control was dealt another blow last week in a highly critical report issued by the Radio Technical Commission for Aeronautics (RTCA).

The RTCA report resulted from the third sequential probe into this area in less than four months. It followed a similar analysis made by the Air Traffic Controllers Assn., and a current review of FAA's use of controller manpower being conducted by the General Accounting Office (AW Age, p. 49). FAA has not replied to either the ATCA or RTCA critiques, but has been conducting its own study of controller utilization for the past several months.

General theme of the RTCA report, the second it has made on air traffic control since 1948, is that FAA's administration of the system has been inadequate to match the growth of traffic, despite the agency's ever-increasing budget for that purpose. Rather than add any further deterioration of the system while lengthier to current projects are completed, FAA should take immediate steps to improve the system through a more efficient use of existing personnel and equipment, the organization and

improved training programs and well standards for air traffic controllers. An increase in personnel, strategic and improvement in existing radio equipment were the most significant items among the more than 70 recommendations made by RTCA.

Based on the RTCA's view of the air traffic control system over the past eight years and FAA's research and development progress were particularly

criticized by the commission, on grounds that the use of scientific methods of the profession to the increased use of the system.

RTCA noted that FAA's budget has "seen through" from the \$335.5 million level of 1955 to \$772 million for Fiscal 1964. While it recognized much of the increase was caused by the addition of high-powered equipment, RTCA said "it is a matter of serious concern, however, that these capital investments show no tendency to produce the desired in the form of reduced costs" such as per aircraft or aircraft mile.

Between 1955 and 1960, FAA cost per aircraft mile increased 244.7% while each aircraft's expenditures increased only 10.6%. Annual aircraft miles during the period increased 29.1%. FAA costs per aircraft mile increased 223.9%. In a further comparison of air traffic agencies with FAA, only the report found that from 1957 to 1962, when total UPR operations in-

creased only 35.3%, FAA costs rose 184.9%. Total airport operations increased 44.8% from 1955 to 1962, whereas FAA costs rose 479%. RTCA stated.

In a breakdown of costs of typical equipment needed for an air traffic control, the organization noted that the expense for such items as a terminal VOR or \$120,000 in a direct effect on pay how many personnel and can be acquired within FAA budget.

"There is some question that excessive specifications have at times been used," the report said.

While RTCA did not elaborate on this subject, a similar question was asked recently by S. B. Postle, transportation manager of the Air Transport Assn. (AW Age, p. 50).

Partially motivated by the use of more "off-the-shelf" equipment, however, as opposed to FAA's current practice of having equipment built to suit specific situations. He pointed out that several states that purchase and install their own equipment are able to offer a factor for the almost identical type of equipment. As an example, he noted that FAA routinely budgets \$15, 800 for a T-19C jet carrier, whereas the State of Nevada buys and installs a similar jet for only \$12,500.

Part of the difference in cost arose from the fact that FAA often provides unnecessary base and cockpit buildings for the installations under contract contracts. Postle said. He contended that FAA could stretch its dollar farther by first making certain that such an elaborate installation is justified in relation to its expected use, and then contracting the entire effort as a single contract. In this manner, he said, Minnesota has been able to purchase land for the first VOR equipment at a cost of only \$10,000 compared with FAA's estimate of \$15,000.

RTCA also criticized FAA on which FAA and its performance. The GAO will have a report on potential \$150 million by the end of Fiscal 1964, even under more critical from RTCA.

FAA has failed to provide any evidence of long-term planning in its air traffic control, the RTCA said. It also noted that the agency's research and development projects the organization changed. Refusal of reports by the public has often been delayed months, and even of the information that is released is not adequately supported by evidence. Despite a limited degree the industry has not been afforded the time needed to conduct a detailed discussion of solu-



Alitalia to Use MB.326 in Pilot Training

Alitalia, Italy's national airline, has purchased several Macchi MB.326 jet trainers due to the aircraft's flight training base at Trapani in Sicily. Alitalia has been forced to implement its pilot training over the small volume of aircraft. The Italian airline services has not been finishing enough pilots. One hundred thousand pilots are training at the Alitalia Engineering Center at Pisa.

vidual projects, according to the report. The lack of sufficient and timely reports has made it difficult "to not impossible" for the industry to determine exactly what recommendations to effect on to justify the value of the various projects, RTCA said.

Noting that the present air traffic control system, RTCA found it has become so complex that it constitutes a hazard for the user, who is faced with a wide variety of rules, procedures and regulations. An effort to simplify the system should be undertaken the group said.

As a result, the workload has been increased for the pilot in a point where it is now a "critical problem" in force law to comply with a variety of extra navigation and communication tasks in addition to doing the actual, RTCA said. The air traffic control system should have in its goal a reduction of the cockpit workload, it added.

As the rate of air traffic increases, so too does the workload of the air traffic controller, who must handle added procedures that complicate his task, the group said. To keep the workload at manageable levels, RTCA said, it is forced to reduce his area of control as the number of aircraft per square mile is increased.

Results of this process, which has been intensified by the trend of sub-jobs in that the controller has a new task to analyze the traffic, make a decision and act on it, the report said. In addition, his task is complicated further by the necessity to operate aircraft controls and controls that divert attention from the controller's primary function.

Scope of the controller's job also has been expanded to include extra duties such as air traffic control, which is not directly related to traffic control, the report said. RTCA cited further reports in pilots as an example and urged that FAA examine its controller workload to ensure that all tasks are for the primary purpose of air traffic control.

Point to World War 2 controller operations were thoroughly stressed, for high qualifications and training to handle actual related problems. Since then, thousands of new controllers have been added and the job becomes more complicated training. RTCA said, noting that a "tremendous deterioration in the proficiency of the finished product is quite discernible."

With the rapid expansion of data required of today's controller, more can should be given to the selection and particularly the training of the individual applicants. The report stated RTCA suggested that FAA develop a special aptitude test for controllers, plus some intensive radio-navigation course at the agency's academy at Olathe, Kan. City, plus conferences, upgrading, seminars and computer courses.

FAA has recently refused a propo-

sition to use traffic control course at Olathe City in favor of more active "on the job training" at towers and control centers throughout the agency's seven regions.

There has also been some dissatisfaction among controllers over a planned job classification change now being prepared by FAA under a former agreement with the Civil Service Commission. Grade classification, per se, which are presently set under a traffic formula for lower controllers, will be changed to combine the present first levels of traffic volume into only three. FAA contends the change will have virtually no effect on controller pay, but the Commission contends that it will raise rates.

RTCA urged that FAA extend its program of positive control at airports in the highly densely traveled areas such as New York, Los Angeles, Chicago and Washington, D. C. The organization did not define any dimensions for the controlled area, but said that the control should be established over at least several miles. Right back, on some control segments below 10,000 ft. A variation of this idea has been used by FAA at Atlanta, Ga. for some time and should be continued, RTCA said. The organization also recommended that FAA continue expanding its area positive control over aircraft flying in and above 10,000 ft.

One of the controller's largest problems is the overlap and confusion of aircraft on radar scopes, the report noted. Top priority should be given to organizing these radar with air traffic control, which means positive data control of identifying aircraft. Currently, RTCA said, each radar position should be provided with independent controls for this purpose. In this manner, and aircraft information would be displayed adjacent to the radar target.

Western Seeks Fare Reduction

Los Angeles—Western Air Lines has filed intention of reducing air fare between Los Angeles and San Francisco, a move which probably will intensify its pressure into other Pacific Southwest Airlines over that important route. (AW Age, p. 40).

Western wants to reduce the fare on Los Angeles-San Francisco aircraft last head Elatus to meet the \$15.10 split one-way fare charged by PSA, a California interstate airline, which operates on helicopter flights on San Francisco-Los Angeles-Sage route. Western's fare change will go into effect Aug. 1 unless imposed by the State Public Utilities Commission, which is authorized.

Western's fare law will apply only to interstate commerce. Those connecting between California to the other states that fly Elatus flights each week between the two cities will have to pay the present \$15.10 fare. "We're leaving the door to meet the PSA competition," a Western official admitted. Western also has 13 Pacific Douglas DC-6B flights weekly between San Francisco and Los Angeles carrying 51-41 one-way in addition to its Boeing 720B helicopter flights daily each way of higher capacity.

PSA is the No. 1 carrier between the two cities, while Western was a close second. PSA added a sixth Elatus to its schedule in late June and now operates approximately 260 flights weekly, one-way and return.

B-62 Progress

More-than-halfway are now test flying three prototypes of the B-62 jet transport in the certification program for Air Force but this one not using the on-line planned his production models of the aircraft.

The B-62 (NATO code: Clans), which made its initial flight in January, 1963, is now in the final stages of development. The aircraft is now being tested by the Air Force at Edwards Air Force Base, Calif. The aircraft is now being tested by the Air Force at Edwards Air Force Base, Calif. The aircraft is now being tested by the Air Force at Edwards Air Force Base, Calif.

Production version which Aeroflot is developing for delivery will be produced by the Kuznetsov turboprop engines in still under development. They are running as good but not yet have not yet begun flight tests.

The two-ducted turboprop will be modified from the present configuration to have the turboprop engine, but there will be no other modifications.

Concorde Tests at U.S. Facility Approved

By L. L. Dety

Washington—U.S. Air Force and Federal Aviation Agency last week announced approval of British Seaboard's request to fly the Concorde Test Center to test the Olympus Mark 591 engines that will power the Concorde supersonic transport (AWJ July 8, p. 20).

The decision was made to permit crash tests on the engines between the Anglo-French Concorde project team and the U.S. in the development of a supersonic transport and, according to one top government official, a better opportunity to achieve the program around supersonic development. The Anglo-French Concorde project team, led by the USAF test center at Tullahoma, Tenn., is the "only place in the world" where British Seaboard can test the Olympus engines.

Meanwhile, the U.S. position in the supersonic transport race seemed divided last week with no tangible signs that a U.S. project would be launched in the near future. Thus far, only the American Airlines and Boeing Administration circles, however, had not Avianet Administrator N. E. Hall has rejected the proposal. British officials have indicated that the U.S. is in the Concorde project primarily because the group has "no test facilities, no fabrication, no transportation."

On the other hand, the decision to grant use of the Arnold Test Center to Seaboard-BOAC Cargo Pact Set

Washington—Seaboard World Airlines, British Overseas Airways Corp. last week signed an agreement for operation of Seaboard's Canadian CL-44 helicopter air-cargo transports on the British carrier's transatlantic routes. The agreement, the result of an agreement with Lufthansa German Airlines, two years ago in that BOAC will lease a plane from Seaboard as well as purchase "blacked" space on a plane operating the route in Seaboard. Under the Lufthansa agreement, the German carrier recently leased a pre-determined volume of blacked space on a daily cargo CL-44 flight operated by Seaboard from Frankfurt to New York.

The U.S. airforce airline has a similar agreement with Swissair on a weekly smaller scale, with the European carrier purchasing only 2,000 lb. of space on a cargo-week flight.

Seaboard created the "blacked" space, which is a means of creating an abundance of traffic on

existing developments, commonly last three years as immediate cargo last would rush to enter Concorde. The American's general cooperation on the North Atlantic route, British Airlines, stated only that it was having a close eye on the Concorde program.

A number of airline officials and government officials were privately discussing the first American order and the impact of the Concorde's impact on the Concorde in "publicly" government officials, including that the construction of the supersonic transport at the test center is in transportation, project. One official stated that, while the Concorde is a "good business" and "good publicity," the construction is "cost-prohibitive" and it is likely to get bogged down under the past arrangement.

He added that determination of whether the U.S. proposed project or the Concorde is superior is still a "pending" project. A representative of the British-British test last week admitted that basic materials requirements for the Concorde, including the following data, a state-of-the-art judge would a temporary restraining order on grounds that the aircraft would cause "irreparable damage" to the airline.

A local court blocked the plane and it was released to the U.S. The U.S. marshals have contacted the plane again and had it moved to an FAA airport, but later the same day, an unknown party, claimed with the original restraining order, prevented the plane from the airport once again to land.

The airline promptly resumed service with the reoccupied plane, and British President William E. Wood said in the plane that night, presumably in order to ensure that the aircraft project would be available for subsequent review. Here is a chronology of events before the settlement with the FAA.

On June 28, Federal Judge John J. Siragusa, presiding in Manhattan, has scheduled hearings on a Civil Aeronautics Board petition for a restraining order and temporary injunction preventing the airline from flying the plane over the U.S. jurisdiction. The FAA said that it is the current agency for international aviation on its international service and it then met an emergency order. Peace also scheduled hearings for the following day as whether the case should be decided by the court's own order.

Federal Judge John J. Siragusa, sitting for Judge Jones, postponed hearings on the same case until he could hear himself with the case. U.S. attorneys delay in further action pending the court's decision.

Wood continued to fly sporadic scheduled, but announced on July 2 it would discontinue operations "voluntarily" rather than risk FAA discipline.

Revised Seaboard was based, in part, on the opportunity such testing will give U.S. engineers to observe what the Olympus engine "can or cannot do." Technical information on the development of a U.S. supersonic transport would be available to Seaboard and British Airways Corp., joint manufacturers of the Concorde, through the exchange of data within NATO.

U.S. Development

U.S. government officials deny the U.S. project, once it is started, will move faster than the Anglo-French project. One official stated that, while the Concorde is a "good business" and "good publicity," the construction is "cost-prohibitive" and it is likely to get bogged down under the past arrangement.

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Lone DC-40 Plays Short-Haul Role In Island Airlines-FAA Struggle

Honolulu—Island Airlines, a certificate holder, in a career, last week, appeared to have ended its struggle with the Federal Aviation Agency after the two agreed to a temporary restraining order on grounds that the aircraft would cause "irreparable damage" to the airline.

Later last week the airline, which has been operating international flights in the face of FAA orders to suspend service until meeting its obligations, scheduled (AWJ June 10, p. 31), spent over \$100,000 to fly the aircraft, monthly, in a temporary restraining order on grounds that the aircraft would cause "irreparable damage" to the airline. In addition, it agreed to discontinue service until the certificate was granted.

Last last week, FAA issued the order \$100,000 in fines and ordered service of the 79-passenger aircraft. The plane was turned to a storage area on grounds that the U.S. aircraft project would be available for subsequent review. Here is a chronology of events before the settlement with the FAA.

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against flight crews of the airline.

On July 3, Western denied the government's request for a temporary restraining order and injunction to ground the airline and set Aug. 3 for hearings on the government's request. The airline based on FAA compliance. The airline resumed operations.

Wood and his attorney, Frank D. Puffer, flew to Washington on the invitation of the Justice Dept. Following discussions in Washington, the airline and the U.S. attorney's office in Hawaii agreed an agreement that, as late last week had brought the FAA island back to an end.

Under the agreement, the airline agreed to file plans, procedures against island upon payment of the fine until most of the \$2,000 fine. This was done and the FAA agreed to "temporarily" provide the airline's application for continued operation, subject to the FAA's approval.

If the federal court order should conclude hearings, which were to begin July 15, in denying a CAB petition for preliminary injunction against island on the "ground" that island is not subject to the jurisdiction of the CAB.

Thus, FAA would possibly issue a certificate authorizing the airline to resume scheduled operations under Part 45 of FAA regulations. Island has three scheduled services, each with a daily flight. When the certificate is granted, island will launch full-time scheduled operations in direct competition with two scheduled carriers, Aloha Airlines and Hawaiian Airlines. Aloha, meanwhile, is assessing what the impact of service conducted by island during the past two months has meant a 6% drop in air traffic and a cut in its scheduled to be applied for additional federal aid.

New York—Vital Flights Air has successfully negotiated a two-year contract with American Airlines, granting the airline the right to operate scheduled flights on the airline's routes. The contract, effective until July 9, 1985, was termed "a complete relief" by an Air Line Pilots Assn. spokesman, who stated that the agreement will ensure benefits and work-life changes.

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A spokesman for Allied Pilots Association, the "best in the industry" will take the rest of the industry down to five years to catch up. An American Airlines spokesman and the company was awarded with the contract. Total cost to the carrier under the new pact is approximately \$7 million annually.

The contract included three major provisions: a pilot's salary for a month in which he was scheduled to fly 16 days would last only 10 days; a 10-day leave; Annual sick leave is increased from 14 to 16 days, with a total accumulation of days increased from 84 to 126.

London—Vickers Aircraft last week was making final arrangements to fly its largest VC10 transport to New Zealand next month for a series of demonstrations.

According to present plans, the VC10 also will be demonstrated in New Zealand and Australia, although the final requirements on a plane type of the two models. The VC10 has been demonstrated only in Great Britain and at the Paris Air Show.

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Maytag Urges Mach 2 SST Development

Maytag—Building a Mach 3 transport would be a "realistic and somewhat delusional" to the U.S. aerospace, says the National Aeronautics Administration's Chief of Research.

Maytag said that the U.S. push ahead with development of a subsonic Mach 2 transport and abandon the attempt that greater speed, capacity and longer range make for a better aircraft.

This philosophy is only good in a pinch, Maytag said and "now out at the margin in world market transactions which we are moving to reach to maintain." Now that we are on the threshold of important commercial transports, when speed is measured in Mach, we are discovering that decisions to race, surge and spend is a waste of money.

By concentrating itself to a Mach 3 transport, the U.S. will have an aircraft suitable only for long money operations such as transatlantic and transpacific routes, Maytag said.

On the other hand, Mach 3 aircrafts, only three—Pan American, Trans World, and Lockheed—could probably use a Mach 3 transport advantageously, he noted. United Air Lines might possibly use a Mach 3 aircraft on its West Coast-Alaska route. Maytag added, but it is "certainly not likely" that United would have such an expensive aircraft for use on only one segment of its vast system.

Mach 2 Market

"If the United States does not develop a Mach 2 aircraft, American trunk carriers will be forced to turn to other air manufacturers," Maytag stated.

U.S. government and industry support of a Mach 3 transport on the grounds that its speed, range and capacity will make a Mach 3 aircraft obsolete before the latter is built is a miscalculation, Maytag said. "The aircraft manufacturers were made to look like Luddites when it found itself in competition with jet jets," he said. "The fact is that it did not become obsolete as well as a Mach 2 aircraft," he said.

Electric equipment on such routes as Washington-Miami and Los Angeles-San Francisco has shown that the passenger air service can be using the direct flying aircraft in preference to the jet jets, Maytag said.

The same problem would exist on domestic routes if a Mach 3 transport were used rather than a Mach 2 aircraft, he added, largely because of the takeoff and approach distances needed to customer demand from near Boston-Maryland and that the present, expanding concept, "The Mach 3 has to climb 32 mi. before it can go into

supersonic cruise." To achieve 3,000 mph speed, the plane will still have to climb gradually in order to open the sonic boom—the overpressure-outlet system.

"In the time the aircraft has leveled off and is flying at cruising range, the work will begin on a much more complicated device. The plane cannot tip toward the ground, or else the mist shower of angles will lead the aircraft into crashing a severe boom."

At a 12-mi. altitude the plane once again will begin a climb in a horizontal track. By the time it has reached a cruising speed of about 3,000 mph, it will then make a gradual descent in its destination airport," Maytag said.

"The Mach 3 airplane is comparatively less firm, more expensive, ground clearance above 33,000 ft. Thus, while the Mach 3 airplane is still climbing at a subsonic speed, the Mach 2 is streaking along about 1,100 mph."

Transcontinental Flight

On a 2,500-mi. Miami-Los Angeles nonstop flight, the Mach 2 aircraft would arrive first, Maytag said. If both aircraft could take off simultaneously and maintain cruise cruising speed, and both stop instantly from cruising speed, the Mach 3 aircraft would arrive in Los Angeles 25 min. ahead of the Mach 2.

In reality, the Mach 3 aircraft would take the Mach 2 transport "somewhere over Texas," Maytag said. In the vicinity of Albuquerque, however, the Mach 3 aircraft would have to begin complicated descent procedures, and the Mach 2 would again prove ahead of Mach 3.

"By the time the Mach 3 aircraft lands at Los Angeles International, the passenger aboard the relatively slower Mach 2 will have a half hour to pick up their baggage and be gone," Maytag said.

Maytag also questioned the cost of

developing a Mach 3 aircraft, which he said is estimated at \$152 million.

"If the development cost of a Mach 3 aircraft are to be added up to the amount, who produce the aircraft," Maytag said, "the sale price of each aircraft will approximate \$18 million. Its potential sales are so limited that a percent cost could reduce far exceed the \$18 million by as much as half again."

In comparison, Maytag pointed out that the selling price of the Anglo French Mach 3 transport, the Concorde, currently is estimated at about \$1 million—half more than the \$6.2 million price of a Douglas DC-8F jet transport.

The government's attitude that it must undertake development cost of a Mach 3 transport leads to "another more dramatic position. The letting of low interest loans has proved true and again that the product of an immediate is inevitably better when it competes with a similar product from a real manufacturer. The factor at stake is the undercapitalized aircraft industry in thought with uncompetitive factors," Maytag said.

It would be better, Maytag said, with either a Mach 2 or Mach 3 aircraft to have more than one manufacturer's proposal entered "at least through the prototype evaluation [rather] than to save money with a chance of failure."

Program Cost

The problem, Maytag pointed out, is that no single manufacturer in the U.S. has sufficient capital to undertake the cost of a supersonic transport program, yet under the free enterprise system, these development costs should be included in the price of the aircraft.

"Such costs are prohibitive and the only possible answer would be continuing means of our current until we reach a positive development before operating on a nonproductive nonproductive basis," he said.

The largest answer is for the U.S. to concentrate on developing an economical, efficient Mach 2 transport if it is to be the result of a manifestation of commercial aircraft, Maytag said.

"On a Mach 2 aircraft in American development, the development costs are not so great but that they are not absorbed by private industry," he noted. "We will be required to accept an even loss with the French and British governments, but we will have a tremendous advantage because under the free enterprise system there will be no real setbacks within the United States."



First World Airways Boeing C-Jet Flies

World Airways Boeing 707-120C jetplane, assembled as passenger transport takes off from Boston, Wash., Municipal Airport on its initial flight. The new "C-jet" is fast of three aircraft will use as transport on regular passenger operations and transport cargo service on Airborne. Flies only 35¢ economy-class passenger to \$10.00 in air cargo.

Program to Soundproof Houses Near London-Heathrow Rejected

London-British government has rejected a program to soundproof houses in the vicinity of London-Heathrow airport. This was one of several recommendations in a report made by the Wilson committee on the problem of noise, set up in 1960.

The committee, headed by Sir Alan Wilson, in a report to Parliament, stressed that noise to which people near London airport are subjected is more than they can be reasonably expected to tolerate, and added:

"We are convinced that the degree of exposure to noise in areas close to the airport will not be naturally or directly reduced, and will give a good deal of stress and annoyance to some people now." These recommendations included:

- Reduction of noise of individual aircraft in new types using two engines.
- Provide houses with the heaviest of the airport with better insulation against noise.

The committee said that a reduction of 7 points (percentage) noise level in the evening peak noise level of aircraft is required to affect the increased number of jet movements reported at the airport between 1951 and 1970. The only possible way, in which this could be achieved is by engineering improvements in engines and aircraft, the committee added.

Ministry of Aviation presently is spending about \$12 million a year on research and development in the vicinity of noise. Much of the work centers on

engine noise suppression at the National Gas Turbine Establishment.

At London Airport, operations are limited to a noise limit of 110 points and are not allowed to take off after midnight unless they have given a high degree of compliance with the noise limit. A number of services have been approved at night for aircraft which have demonstrated that they can operate at noise levels not exceeding 105 points.

Ministry of Aviation has allowed operations freedom to achieve the noise limit in use, which can, subject to the main consideration of safety. Two techniques are present in use:

• Full power used on takeoff to give a steep rate of climb until the aircraft approaches the final climb-up angle. Just before this point, power is reduced to

the minimum thrust needed to maintain a steady climb with safety. Full power is added after passing the area. • Reducing total fuel to enable the air plane to climb to a height where the noise level will be within the limit before the power is added. This method causes a later step in climbing, at considerable cost.

Currently, Ministry of Aviation is monitoring about 90% of all aircraft from London-Heathrow, but the Ministry does not intend any expansion which will automatically increase the noise level of all aircraft.

The committee has rejected a suggestion that all aircraft be slowed during ground operations. The group noted that this restriction by noise suppression would not counterbalance the difficulties caused for the airport authorities and the aircraft operations.

In response to Parliament, the committee emphasized the commercial aspects of noise suppression at London, noting:

"If, in the interests of noise reduction, restrictions are placed upon air line operations at Heathrow which have such serious economic penalties that the use of the airport becomes unprofitable for them, then air transfer from airports in other areas around a substantial reduction in the air services to Heathrow, which serves London and the whole country, would have grave deleterious effects on the country's economy."

In reference to soundproofing houses bordering Heathrow, Lord Hailsham, the minister of science, said the government is not satisfied that the solution at Heathrow parties adoption of the recommendations and that he believed it would be impossible to limit growth to houses near airport.

Austrian Load Figures

Vienna-Austrian Airlines' preliminary results show gains of 15% in tonnage and 18.5% in passenger carried 27.7% in freight and 23.5% in mail carried during the first half of 1961 over the same period last year.

Tonnage totalled 5,686,260 in comparison with 3,213,139 tonnage during the first six months of 1960. Passenger carried were 76,344, freight loaded 415,515 U.S. ton, and 198,515 U.S. ton during the first half of this year compared with totals of 61,290 passenger 317,3 tons of freight and 192 tons of mail for the 1960 period.

What is a fan-jet?

The American Airlines fan-jet story

A jet is propelled by the thrust of its engines.

But this thrust comes from a very hot exhaust, and hot air is thin air—a little like a lightweight's punch.

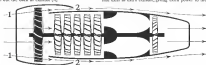
The fan-jet makes a sort of heavyweight out of it.

This engine, which American Airlines helped develop, takes



[Above] Ordinary jet doesn't use the extra air through an exhaust side (1). This extra air is partly compressed (2). This extra air is then cooled (3) and then exits the back as exhaust (4).

[Below] Fanjet draws in extra air through an exhaust side (1). This extra air is partly compressed (2). This, and cool air drawn through side exit as extra exhaust, gives extra power to the plane.



in twice as much air as ordinary jets and gives the thrust twice as much body. The result is 30% extra power—so much that the plane itself had to be changed.

This was the birth of our first Astrojet in 1961.

Astrojets take off and climb faster, fly more quietly, and use the fan-jet's extra power to help get you in on time.

Only 2 airlines in the U.S. have fan-jets on every jet they fly. American and Western.

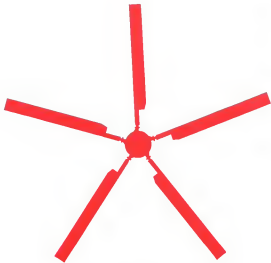
American has 64 fan-jets, 41 more than any other airline.



Caravelle Super A Takeoff Performance Shown

Takeoff performance of the Sud Caravelle Super A transport, powered by General Electric CJ605-23C turbofan engines, is demonstrated at Le Bourget Airport during the Paris Air Show (above). At a maximum gross weight of 114,540 lb., the airplane, with its double-slotted flaps extended 10 deg., requires 6,000 ft. for takeoff. The lightly loaded airplane at Paris was able to improve on this performance as reflected by the steep climb after landing (shown) and by the amount of money remaining. Note given on landing edge of wings at the 100th (LW June 10, p. 45). In the Caravelle Super A with at the end of the money (below), the de Havilland Trident, powered by three Rolls-Royce Spey turbofan engines, starts a low-level pass over Le Bourget.





Cost-cutters

With a helicopter, it's simpler to cut costs, cut downtime. Sikorsky does. The new twin-turbine S-61 is authorized 1,000 hours time between overhauls. That's long enough for it to make four trips around the world.

This economy is another aspect of Sikorsky's leadership in vertical flight. Through such *peace preferences*—and continuing progress—Sikorsky is creating a new world of mobility.

Sikorsky Aircraft DIVISION OF UNITED AIRCRAFT CORPORATION

STAMFORD, CONNECTICUT

U
A



AERIAL VIEW OF O'HARE from west shows terminal complex and 8,700 car parking lot. Present terminal buildings and smaller extension are at right. International arrivals building (left) is being completed. International passenger new arrival at top. Old building at far left. Terminal bus loop (right) with one-way system construction. Housing and transportation plant building is at the parking corner and building in upper right house cargo facilities.

Major U. S. Airports—Part 5a

O'Hare Walking Distance Stirs Criticism

By Robert H. Cook

Chicago—Chicago O'Hare International Airport, the most heavily used air terminal in the world, is caught in a pincer's grip of its own creation.

Management's pride in O'Hare's efficient handling of traffic volume that would overwhelm almost any other terminal is being severely wounded by the continuing public desire to "do something" about the walking distance at O'Hare, designed as one of the first, fastest and largest terminals to serve the public in the jet age.

And the problem will undoubtedly grow worse as traffic continues to increase. Last year O'Hare handled 13,298,709 passengers, and for the first four months of this year recorded a 17.4% gain with 4,711,166 passengers. This annual total is about 10% more than that of New York International Airport and twice that of the Los Angeles International Airport.

Thus, it seems that the pleasure man O'Hare's finger design is too often weighed out for criticism is that the terminal is used by more people than any in the world. Its expansion to maintain is therefore higher and complaints are reported to increase in proportion to traffic growth.

Recently, O'Hare's design reg-

arded a novel way to avert this problem, but was slowed by the dilemma of the airlines. The design first choice was a control terminal building with mobile lounge transportation to the aircraft, to cut down walking distance.

This was rejected by the airlines on the theory that it would only complicate the walking problem for interchange passengers. Today, a local service airline passenger transferring to a transline may walk anywhere from 4 to 8 mi. in making a transfer, unless the local and trunk carriers are located in the same finger system—such as North Central and Lake Central, which are in the same Y shaped finger as Delta and American Airlines.

A second choice for a design similar to the Los Angeles satellite building system was almost accepted but finally was rejected on grounds that it failed to provide sufficient space for airline operations offices.

O'Hare's management is aware of this problem but lacks an acceptable solution as hampered by the high costs involved. Installation of moving walkways through the terminals' two fingers, some of which are two city blocks long could cost an estimated \$3 million and maintenance costs alone would approximate \$300,000 a year.

The airport also has explored the use of passenger cars propelled by a moving belt to connect its two present terminal buildings with a new international terminal now under construction.

In any event, the cost of any proposed modification will be passed on to the airlines, in the form of increased landing rates, or to the taxpayers.

Previously, O'Hare's explosive traffic growth has provided the close support of the airlines and produced a high degree of local stability for the airport. The airlines underwrite the \$140 million in revenue bonds required for construction of the 7,200-acre facility. As part of the past agreement, the airlines have a sliding scale landing fee designed to cover bond principal and interest payments, whether traffic is high or low. The fees to exceed each six months, and have fluctuated all the way from 7 cents to 51.25. After retirement of the first bond issue, these fees will be charged to cover only the airport's actual operating expenses.

Because of the steady increase in O'Hare's traffic volume, construction scheduled on the bond issue has been reduced from the original 40 years to only 27 years.

As O'Hare handles only a small number of international flights, a bus waits each flight and delivers passengers to

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never
been
a filter element
that
could
do
all this



**IT'S A PUROLATOR
PRESSURE LOCK FILTER ELEMENT
AND LOOK WHAT IT DOES:**

- Gives you a filter that's absolutely free of built-in contamination
- Boosts top temperature limit of conventional stainless steel mesh filter to 1000°F or more
- Cuts 43% off the weight of an identical filter assembled by conventional bending methods
- Significantly increases available element filtering area
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- Fits any known hydraulic fluid
- Meets requirements of specifications MIL-P-33048 or MIL-P-33015

**Only Pressure Lock will do it,
and this is why:**

Pressure Lock is an exclusive Purolator process of assembling a true-mesh filter medium on tortuous walling, brazing or crimping, and without the extraneous factor that any of these methods produce. Pressure Lock embeds the filter medium in the end caps in an absolute bond that contains no extrinsic other than the metal of the cap and filter medium. As a result, the only limitation on the performance of a Pressure Lock Filter is design stress, pressure, vibration, filtration or weight — is the physical properties of the construction itself.

Present trends are for a nominal 10 micron (absolute maximum particle size of 20 micron) and for a nominal 5 micron (absolute maximum particle size of 15 micron); in capacities of 3, 4, 12, and 24 CFM.

The Pressure Lock process provides, literally, a new standard for filter applications in aircraft and marine, industrial, hydraulic, pneumatic and chemical process industries. Write for specifications sheet SC 960.



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Three United Boeing 727s Await Flight Test

Three United Air Lines Boeing 727 airplanes are shown on the parking lot at the Renton, Wash., Municipal Airport, prior to making their first flights. First 727 is undergoing flight tests at Boeing Field in Seattle, and have logged nearly 300 flight hours. First 727 different are scheduled for later this year.

the oldest terminal building. Cost is \$7.5 million. O'Hare's own attempt to provide first-class passenger facilities on the outer extremity of the 5,800-car parking lot drew numerous complaints. The cost ran about \$4,800, and the service was eliminated after a closer area of the lot was completed. Future expansion plans call for the double-decking of the south portion of the parking lot. Parking volume now reaches 7,000 cars per day.

Because of its sprawling layout, O'Hare has concentrated its best design facilities in a separate modular building located between the two terminal buildings. The cost plus a towing problem for the taxiway, since it crosses a possible movement, making distance of up to four city blocks to the design gate of almost all airlines, with the exception of United and Trans World Airlines. Veterans airline passengers often prefer a less cumbersome motel at one of the small huts in the terminal buildings or in the fingers.

However, there is no doubt that the airport has entered the passenger's needs whenever possible. In terminal buildings can 21 nearly even considerable need, including both, equipped with areas where infants can be changed, ejected and even fed from a own patch, equipped kitchen.

Geography also city design relating to airport structure, lines within the airport area have also been noted. Given anything such form now are given a special and which across them to their former position in the old line airport, a short-haul way.

Public complaints over aircraft noise have deepened considerably through in, causing one of turboprop equipment and a lead acquisition and moving process, which has provided O'Hare with an increase in traffic volume between it and surrounding communities. About 1,500 acres have been added to the airport in the past year, growth, for aviation education programs are under way or planned. To the north, the acquisition

of Riverwood Airport, a small private field, will provide an alternate approach area for the proposed 4,250 ft. extension to the existing 7,145-ft. Runway 23R, 19L. Cost of the project is estimated at \$1.6 million. To the west and south, the airport boundaries have been extended up to the Chicago and Northwestern Highways and the Chicago, Milwaukee, St. Paul and Pacific railroad lines. To the east, the project has, generally follows the Tri State Toll Highway and the Minneapolis-St. Paul and Saint Louis Main Railroad.

In addition to Runway 23R, O'Hare has 21L, 14R, which is 11,680 ft. long, 23R, 31, at 7,400 ft. 23-4 at 7,500 ft. and 30-11, 5,300 ft. All except the latter two have conventional landing system. Before the end of this year, both will be let for a sixth runway, oriented in an east-west direction. It will be designated Runway 36-37L, and will be 10,000 ft. long. Cost will be approximately \$6 million. The airport's master plan also call for a 5.5 million expansion from this year to general landing of the Northwest Expressway, to connect the terminal ramp area with the terminals of Runways 27R and 27R.

Construction of the new international terminal building should also be finished this year. About 52 million is being expended to bring the five of the second building facility with the terminal building, and the new building, to the north of the new building will be situated for international departures. International arrivals are now being handled in an old Quonset-type building past beyond the finger.

Relocation of O'Hare's traffic control tower is also planned for the near future. The present tower is located atop the Quonset-type D-4000 between the new international building and second building. It will be relocated on the parking lot area directly across the entrance to the terminal building for a better line of sight of the field area.

Gate handling facilities will also be expanded this year with the addition of

two new cargo buildings. The new cargo area is on the east side of the new terminal.

O'Hare's traffic acceptance rate is about 62 aircraft per hr., but it has recorded days when more than 100 arrivals and departures have been made in 1 hr. To prevent any ramp clutter which could impede the acceptance rate, most of O'Hare's management and supervisors personnel are equipped with portable radios, which need to coordinate a smooth flow of ground traffic.

Much of the airport's success in handling ground traffic must be attributed to its semi-controlled facing system which virtually eliminates the need of facing lines to merge with the parked aircraft. A 50-track, South Long Island field, is located on the western perimeter of the ramp, and is connected to five "sideline" parking stations serving each of the finger projects. Two lines to these stations, plus the individual aircraft facing in lanes, is controlled from the fuel farm control.

In the wake of O'Hare's growth, the old Midway Airport has virtually been abandoned with the exception of air scheduled before and general aviation flight. Contracts will be effect for most of the former scheduled runs of Midway, as, due to space need, and still there are no concrete plans to redevelop the facility area, since it is essential for the aircraft are. One reason is the world's largest airport Midway handled only 699,000 passengers last year, with more than half of this total accounted for by non-scheduled and general aviation carriers. Passengers to the latter airports continued. In revenue this year, even as the scheduled airline passenger demand for April, the total passengers handled were 352,000. Scheduled airline passengers accounted for only 3,906 of this total.

(The note with the area on cargo U.S. as terminal will be a report on New York International Airport. It will appear in a subsequent issue.)



Landing strip for a supersonic fighter

Powered by the Bristol Siddeley BS 106 vectored thrust engine, the Hawker P 1134 strike aircraft combines supreme performance with vertical take-off and landing ability. The aircraft is now being developed as a Hawker replacement for the Royal Air Force.

The choice of the BS 106 for the P 1134 follows closely on the success-

ful trials of another Bristol Siddeley powered VJ1076 engine—the Hawker P 1127. The P 1127, powered by the Pegasus vectored thrust engine, has been flying for more than two years and will be followed by the VJ1076 strike aircraft. With more than seven years' development experience in vectored thrust engines, Bristol Siddeley are world leaders in this field.

Bristol Siddeley Engines Limited,
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**BRISTOL SIDDELEY
SUPPLY THE POWER**

SHORTLINES

► **Alitalia and Marco Polo Tours**, a travel agency, have sold 20,000 air round the world trips to 135 members of the Watch Tower Bible and Tract Society at a total cost of \$5.445,000. Tour will include 37 scheduled airlines—200 hotels and numerous bus and train lines. Tour will take six months and will cover 28 cities in 23 countries.

► **American Airlines** reported a 13% gain in revenue passenger miles in June compared with June, 1962. Carrier attributed the increase to "strengthen in the general economy" and "the effect of promotional fares" on sales.

► **Continental Air Lines** today completed the transfer of its headquarters to Los Angeles after more than 25 years in Denver. Total of 135 members of the airline's executive and administrative staffs moved into the company's new \$2 million building at Los Angeles International Airport.

► **Flying Tiger Line** says its expansion program of flying consolidated shipments of parcel post, between Los Angeles and four eastern markets, has generated an additional 1 million lb of air freight business during the six months the program has been in operation.

► **U. S. Commerce Dept.** has reduced the guarantee for the Avarani Loan Guarantee program from \$7500 to 25%.

► **Delta Air Lines** will begin Carrier 800 service into Jackson, Miss., in October, giving Mississippi the state's first jet service.

► **Five American World Airways** has reported a record 13% increase in the number of passengers carried on its North Atlantic route during the first 23 days of June compared with the same period last year.

► **Spangco Engineering Corp.** will install jet fuel hot equipment and main technical facilities at Addis Ababa airport under a contract to TWA for Ethiopian Airlines. TWA provides transportation assistance to the African airline.

► **TWA World Airlines** has reported a 37% increase in revenue passenger miles flown in June compared with the same period last year. International revenue passenger miles climbed 40% in June, while domestic traffic rose 16%.

AIRLINE OBSERVER

► **Deccan** may be made soon by Central Dynamics/Comair as a replacement for the Convair 440 140-440 series and for other twin engine transports. Engineering studies are under way at San Diego and market surveys have been in progress. The airplane would have a trim airframe design mounted at the tail. It would be smaller and lower priced than the Douglas DC-9 BAC 111 or other, larger competitors for the short haul market and would carry 44 passengers—more nearly approximating an Eastern Air Lines requirement for a 50,000-lb. gross weight airplane (AW Nov. 5, 1961, p. 41). Approximately 1,070 of the Convair 440 series aircraft are in operation, about half of them in commercial use and the balance in military service.

► **International Air Transport Assn.** Traffic Conference will be re-opening, may be extended by the agreement by governments to suspend scheduled international fares after this week's meeting of 25 governments in Ottawa. The meeting was convened at the invitation of the Canadian government's Air Transport Board, and is expected to review closely IATA's currently role, considered by many to be the main block to lower fares.

► **Five American World Airways'** plan to extend its present flight from New York, and Pacific Run to Europe and Hawaii (AW July 3, p. 37) is a reversal of former airline traffic structures calling for a reduction in rate per mile as stage length increases. Pan American's fare per mile on the New York Pacific Run route is about 1.6 cents. Under the proposed flight plan, rate will be about 4 cents per mile on the longer West Coast-Hawaii route and 4.5 cents per mile on the still longer transoceanic route.

► **Boeing** received a direct order from the U. S. for an extension of its routes to Los Angeles because of Italy's unwillingness to accept the subsidies that would be exchanged between the two countries. European countries have finally agreed to balk at U. S. demands for the reciprocal trade of such figures.

► **Central Eagle Airways**, the British independent airline, has bought five of the 14 Bristol Britannia 312 turboprop transports planned for all service by British Overseas Airways Corp. The airline has also formed a subsidiary team to evaluate the Victor Super VC 10 four jet transport. Meanwhile, Commonwealth Airlines has ended negotiations with BOAC for five 512s.


► **Imperial Airways** is sending a delegation this month to the U. S. and Britain to select transport planes for expansion of the carrier's operations. The group will study the entire spectrum of medium-range transports.

► **British Air Registration Board** has submitted a report to the Ministry of Air about its much-contested pilot procedures for noise abatement at London Heathrow Airport (see p. 41), after extensive tests using Boeing 707s and de Havilland Comet transports. One of the spots by ARB Chief Pilot A. D. Davis, which has not yet been made public, is that existing regulations are safe but some modifications are needed.

► **Civil Aeronautics Board Executive Robert L. Felt**, in an initial decision, found that the tariff filed by American Airlines calling for reduced fares for former employees is properly discriminatory and unlawful. American requested the first for use of members of the Kew Club, former stewardesses, and Gary Eagle Club, former pilots, in flying to and from club conventions.

► **Five American World Airways** and the Brotherhood of Railway Clerks last week signed an agreement providing for final and binding arbitration in any dispute about the exclusivity of the Railway Labor Act law has been exhausted.

► **United Aircraft Corp.** on cargo shipments during 1962 totaled 9 million lb, or 4.9% of all available tonnage owned by the manufacturer. According to International Commerce Commission's Bureau of Transport Economics and Statistics, less than 0.9% of all freight moved in the U. S. is shipped by air. Statistics apply only to cargo readily transportable by air and do not include such commodities as fuel or building materials.



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Detail of this week's cover photograph that from another angle shows Thunderbirds making steep maneuvers with one F-100 inverted.



Formation maneuvers in first plane elements, above, show steepness and (left) and changing planes in formation (right). Below, two Thunderbirds make extremely low level high-speed maneuvers above the runway as the Thunderbirds make steep maneuvers.



Thunderbirds Introduce New Maneuvers at Paris

USAF Thunderbirds aerobically team performed on both of the last days of the 19th Paris Air Show at Le Bourget introducing several maneuvers that were new to European audiences. The Thunderbirds being North American F-100C Super Sabres, worked in two elements one a four-plane formation that executed precision maneuvers while changing its formation the other two solo planes that filled the voids between the formation maneuvers with a variety of maneuvers close low-level maneuvers. The Thunderbirds used electronic bursts during their maneuvers and faded red white and blue smoke in formation loops and some maneuver patterns.



Sequence, this page shows Thunderbirds making low level passes then banking into loops in opposite directions. Bottom, smoke trails show low level approach as F-100s begin climb into loop. Two top photos show inverted bankovers all top of loops.





Greek F-56 aerobatic team is shown in formation takeoff with smoke F-56. Sides team very closely at La Bourget.

Greek F-56 aerobatic team is shown calling formation approach in landing with gear and flaps down past over the end of the runway.

Greek Air Force Team Displays Precision Formations in First Appearance at Paris



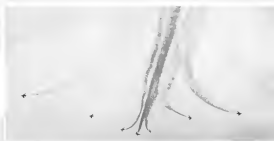
Greek air force aerobatic team displays precision formations with aerobatic (left) and box of diamonds (right).



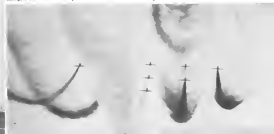
Seven plane formation landing in down past below threshold with nose plane leading first and last plane last to break down.



Italian air force acrobatic team performed one of the most spectacular maneuvers of the 29th First Air Show with the two outer wing men in a Vee formation rolling (above). Note P-51 at right starting roll.



Spectacular beginning of the Italian one-plane double bomb burst (AW June 26, p. 31) is shown with the halfed downward lookaway (top and middle). Eight planes later maneuvered for spread bomb burst with north plane rolling spread through the formation. Kansas pattern shows bottom of formation loop.



Italian Display Features Double Bomb Burst

Nonplus Italian air force acrobatic team earlier last week flew formation pass down runway streamer red, white and green smoke.



First Gemini Computer Model Completed

By Rodrick D. Hibben

Orange, N. Y.—International Business Machines Corp. has shipped the first engineering breadboard model of the Gemini spacecraft computer, a 54-tube unit in Gemini's cabin and docking station, to McDonnell Aircraft Corp.

IIBM Federal Systems Div.'s Space Guidance Center designed and built the computer for McDonnell under a \$4.7-million contract for the inertial guidance system. The subcontract called for 11 spacecraft simulations and two spares.

Digital computer at the heart of a complex inertial guidance system. The station will accomplish the control of Gemini's maneuvers and docking of a manned spacecraft with an Agena D.

Astronauts will navigate from visually displayed commands, presented in the computer, which will signalize Gemini's motion flexibility by giving the crew complete attitude and short control of the spacecraft during each mission (AWE May 11, p. 51).

The computer will be used as a primary or backup system in the following mission modes:

- **Ascent**, the period from launch through orbit insertion of the spacecraft.
- **Readiness**, the period from orbit

insertion through fast and slow catch-up maneuvers, its chase, which provides docking.

- **Orbit navigation**, the period in which spacecraft orbital parameters are being obtained from its solution of a particular landing site.
- **Re-entry**, the period of controlled descent following reentry.

The inertial guidance system, of a ball this computer is a gas, locks up the inertial compass unit, during ascent. The astronaut can decide to switch to the inertial guidance, lock-up unit at any time during ascent should the flight suffer, which indicate pitch and yaw, drift continuously from the inertial position. Depending on the drift rate, the astronaut would manually re-align the drift unit with the ground control station before switching to the inertial guidance system.

A navigation doctrine is shown, consisting of a set of redundant sensors, motion resolution channels and task parameters in both stages and having rates along pitch roll and yaw axes, quantities of fast and second stage, and low voltage (AWE Sept. 3, p. 51). It automatically transfers control of the spacecraft from the radio command unit to the inertial guidance system in the event of a radio primary system failure.



FIRST PRODUCTION PROTOTYPE of Gemini spacecraft computer is shown, lower right, surrounded by special fixture which allows building and test of circuits prior to soldering them into bulk panel.

During ascent, until booster engine cut-off, the computer is operating in an open loop mode. In open loop, it gives automatic guidance and control commands, but doesn't feedback to correct them. The computer estimates the launch azimuth angle, target plane, spacecraft position and velocity. Spacecraft position and velocity are read out every 0.5 sec. The computer gives pre-programmed steering commands to its vehicle until, then the path correctors to coincide the open loop part of the ascent.

Closed Loop Operation

Shortly after last stage burnout, closed loop computer operation of the ascent begins. Using onboard guidance equations, the computer starts steering maneuvers following ascent stage operation.

Ground control uses velocity updates which the computer uses to adjust guidance error.

After maneuvers (i.e., the condition of highest dynamic pressure), the Gemini launch vehicle autopilot gas is controlled by the computer. Second stage engine cut-off for orbit insertion is adjusted by the computer which then calculates the spacecraft thrust for subsequent orbit insertion altitude. Orbit insertion completes the transition from an ascent to a chase mode.

Readiness involves two submodes: fast catch-up and slow catch-up.

After a coast orbit insertion attitude has been established, the mode switch is turned to the fast catch-up submode. At this time, the Gemini computer lies the Agena D target vehicle in 10.75 deg as measured between each spacecraft's longitudinal axes. Cross range correction of 5.5 deg value is then made until the lagging angle is about 10 deg. Target and spacecraft orbital parameters monitored on the period are sent to the computer - the lagging angle approaches 10 deg.

Velocity Changes

Ground computers decrease ascent's incremental velocity changes and thrusting time for readiness. Cross measurements to Gemini from the ground is by voice as an onboard digital command system which accepts beamed coded messages at 1 Hz.

Decision whether to cut to radio view is made as the fast catch-up submode. Decision is based on a favorable fuel requirement, since the astronauts have a thrusting budget they must abide by. If the fuel requirements for

ascent are exceeded, the astronauts must wait.

If the decision to readiness is made, the astronauts will wait incremental velocity changes and thrust time obtained from the ground control station into the ascent data keyboard which is part of a manual data insertion unit. The unit also contains a manual data insertion which verifies information being inserted into the computer as well as adding the astronaut's information contained in the computer memory. An elapsed time indicator provides a thrust time reference for readiness to the crew.

The astronauts depress the spacecraft in the proper orbit plane by drawing three visually displayed general angle outputs from an inertial compass unit to zero. When rolling of the display of general angles is complete, the astronaut pushes a computer start button which activates the computer and its incremental velocity indicator tells the crew what thrust changes must be made as the three velocity axes of the Gemini capsule is further adjusted the spacecraft alignment for readiness.

When the plane angle between the spacecraft and target vehicle is about 10 deg, the slow catch-up submode begins and smaller alignment corrections of about 1.5 deg value are made until the lag angle is about 7 deg.

Radar Lock-on

Radar lock-on occurs at a mean main stage of 250 mi (AWE Oct. 22, p. 72) activating a red light. Radar gas enters a hot position aspect to the computer, which changes at lock on closing the red light circuit. Mode switch is then put in the readiness position.

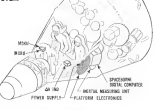
Radar data points are fed into the computer every 100 sec. A total of 24 readings are obtained in the open loop part of readiness. The computer translates the radar points from digital form into incremental velocity changes in the three spacecraft axes—pitch, yaw, roll—and displays them to the astronauts.

The velocity changes displayed are calculated from the first sight and the waveform 15 and 25 under points.

At this time, velocity changes to see decrease would be about 100 ft/sec or less. The astronauts would then pass the computations start button to initiate the closed loop portion of the readiness view.

The computer display board would

GEMINI CAPSULE INERTIAL GUIDANCE SYSTEM



GEMINI CAPSULE's ascent data keyboard unit (MIDK) and ascent data readout unit (MIDRU) positions are shown, above. Initial orbit position of Agena D and Gemini, below, presents 10.75 deg leading angle between longitudinal axes. Visible blank on north capability of Gemini offers some degree of flexibility in maneuvering plan.

INITIAL ORBIT RELATIONS



then become a flight instrument. When the incremental velocity indicators display step, the astronauts regulate thrust to correct the error. Spacecraft attitude errors are corrected manually. After the astronauts have obtained a sense of seven radar points following thrusting, they will use the gyro com-

pass for reference alignment of the space platform.

Attitude and velocity errors are corrected using a digital technique. This technique involves making a series of corrections starting first with gyro correction then smaller and smaller corrections. Errors in attitude and velocity

in every pound of



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PROGRAMMED ROLL: Roll is initiated 520 sec after liftoff. Launch vehicle has 28-deg roll capability clockwise or counterclockwise. Attitude guidance control (AGC) follows programmed pitch command. No roll condition during ascent, right, corresponds to 28 deg/deg roll rate. Roll is begun at about 60,000 ft. Spacecraft maintains rollrate at 60,000 ft.

An automatic landing schedule post 30 deg from impact on the Henderson respiratory suits for the closing upstrut, indicated by radio to be maintained usually by the crew. Visual navigation of the spacecraft would provide docking with the Agena D.

Gemini Emergency Alert

Feasibility of two emergency shut-out modes resulted from IBM's Greenleaf irrigation studies. In the first mode the stations would plot their ground path, select a preferred landing site and determine retrograde trace flow-patterns. Subsequent ERTS-1 orbit passes would provide a chronological tabulation of the spacecraft's position—considered as a celestial body—of regular trace stations. Landing at a preferred site neither was critical nor required in the first mode.

Second emergency mode was synchronous with the radio command and means alongside base instructions for recovery. Semi-optical tracking technique employed in the ground station would establish the spacecraft initial position to target at landing site. Astronaut would then have the option to decide when to go. Further investigations on the implementation of the concepts in orbit are planned to enhance the flexibility of the second rescue mode.

Emergency shut technique for Gt guns will be selected from the more flexible of the two codes.

If overload of the spacecraft should occur, an alternate mode of the automatic guidance system would break its link to the computer, activating a red light indicating a down range error. The guidance system would correct the overload by putting the spacecraft in a soft lift trajectory. Cross range error would be displayed similarly.

Landing footprint at 100,000 ft will be an area about 400 m long and 80 m wide. Re-entry target is expected to be Capricorn, Christ, Tex.

The computer screen "The most advanced computer we have built" by Harry F. Beaming. Gears within an octagon monitor at IBM, weighs 57 lb. occupies 14 in. ft and requires an amount of 94.54 w. power.

The computer is designed as a general purpose device and serial with a fixed point arithmetic operation. Basic element of the random access mem-

destructive advanced memory is a true full-fledged cache developed at IBM. Non-destructive can, flatbed of the memory since power and need for added infrastructure, IBM said.

by stacking 79 planes, each consisting of a 64 x 64 array of cones. Total memory, outputs is 4,096 words of 16 bits per word or 179,744 total bits. Suffix files are 13 bits in length with three affixes in across reading up one increment word. An instruction word corresponds to one affixable, a left a data word can be one or two affixables.

Division between contraction and

51



on target with airlift power...and helping provide the increased mobility a modern Army needs by advancing the state of the turboprop art. Soon the USAF C-130 will carry larger payloads over longer distances, because air-cooled turbine blades will make its Allison T56 turboprop engine more powerful and efficient. Thus our defense posture is more flexible, our airlift capability geared to today's tactical requirements, our Army better able to carry out its mission, because another Allison program is on target.

Allison 
THE ENGINE CONSTRUCTION DIVISION OF
GENERAL MOTORS CORPORATION, INC.

data storage in the various case per minute flexible operation of the computer.

Basic computer clock rate is 5 mc which is used to convert platform generator pulses from analog to digital form. Four pulses are generated from the 5 mc clock rate to provide the reference timing for all of the other computer events.

The airborne lat rate operates at 500 kc and the ocean cycle rate is 250 kc. Adding, subtracting and transferring of information takes 140 microseconds.

Full precision multiplication and division require 650 and 500 microseconds, respectively.

All arithmetic operations can be programmed consecutively.

Mission reliability of the computer—the mean operating time between failures is about 2,500 hr.

A diagnostic subsystem is built into the computer to detect malfunctions during operation. When a malfunction occurs, a discrete signal sent by the computer activates a light on the crewman's control panel. The crewman can act on a manual reset capability to

Computer Characteristics

Inputs	No.
Discrete	40
Incremental velocity channels	3
Control angle channels	3
High speed (500 kc) data channels	2
Low speed (100 kc) data channel	1
Low speed (100 kc) data channel	1
Input and output channel (90 words)	1
IO C power (4 regulated)	5
A/C power (regulated)	1
Outputs	
Discrete	10
Steering command channels	3
Steering velocity channels	3
Decentral display (7 digits)	1
Telemetry channel (21 digital data words)	1
Low speed (100 kc) data channel	1
Low speed (100 kc) data channel	1
Regulated dc power	3
A/C power, regulated different	1
Miscellaneous	No.
Altimeter	words
Fuel tank	40
Acoustic guidance	2,500
Clock/pulse/encoder	2,500
Orbit navigation and events	4,000
Manual input and display	400
Input/output	1,125
Compass calibration	575
Total	22,175
Total available	12,200

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MANUAL DATA INSERTION UNIT, left photo, is composed of manual data reader (extreme left) and manual data keyboard. Unit allows astronaut to communicate with computer. Incremental velocity indicator, right photo, reads 0-999 ft/sec. changes as three spacecraft axes

sway that it was not lighted by a transient type malfunction.

The computer is controlled with four switches located on the astronaut's control panel.

The switches are:

- Seven-position computer mode switch
- Two-position computer off-on switch
- Push-button start computation switch
- Push-button nullification reset switch

Commands in the power spacecraft power section are handled by a regulated d.c. power from the inertial guidance power supply. Regulated a.c. and d.c. from the inertial guidance section power supply are the computer. Power rate guidance and depression of up to 36 mil/sec. and 10 mil/sec., respectively, can be followed by the computer.

The inertial guidance system occupies 11 cu ft, weighs 177.75 lb and needs 448.8 w, almost 10% of Gemini's gross power.

Information from the computer is the following units of the inertial guidance system:

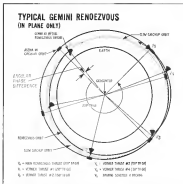
- **Inertial platform**—Computer supplies a 400 cps rotation signal to three sensors located on the platform, enter will and yaw goniometer of the platform. Movement of the rotors of any of the motion over time than zero position- platform signal position causes the output voltage of the rotors to change on each goniometer unit to be shown shifted relative to the reference. 400 cps. rotation. Each goniometer supplies two cps. voltage signals to the computer.
- **Platform electronics**—Outputs from each of three platform sensors are supplied as incremental velocity pulses on two input lines to the computer.
- **Manual data insertion unit**—This unit can insert into and display up to 99 data words into the computer. Read-back of the manual data insertion unit has 10 push buttons, and the astronaut will have to insert seven digits to obtain information from the computer. The

last two digits address the computer and the following five feed in data.

If the operator should insert a request readback of an inserted address, insert more than seven digits or fail to insert an address prior to depressing the insert or readback button, the seven digits displayed will be all zeros to indicate automatic error.

• **Readout rate**—Computer receives line of sight range to target, use of the

vision and measured angle within a 74-deg. maximum error, on value line of sight relative to the spacecraft axis. Gemini is navigated using two coordinate systems—a platform and a spacecraft reference axis. The readout rate is a word computer data in the space reference. The computer must translate all position and attitude information into platform coordinates except attitude changes to be made with the op-



SLOW CATCHER ORBIT following fast circling to Gemini rendezvous mode provides selected alignment of spacecraft and target vehicle when multiple phase difference between the two is 18 deg. or less. Complex rendezvous trajectory in three-dimensional. Spacecraft is expected to pass through many planes before beginning with Agena ID.



With communications links and command positions, free from long range attack, are a part of our deterrent power made necessary to provide a world free of uncontrolled aggression.

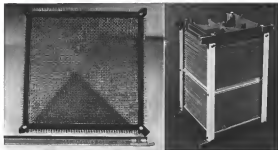
Operational today is a network of flying command posts and communications relays which play a vital part in America's deterrent capability. An important part of this airborne system is the AN/ARC Rb(V) UHF Multiplex Link installed aboard modified B-47s aircraft of the Strategic Air Command by Temco Aerospace Systems Division of Long-Tecoma-Vought.

Temco Aerospace Systems evolved a prototype configuration, installed two in modified aircraft and performed contractor flight test in just 65 calendar days after receipt of the contract and aircraft. This record was made possible through the use of Temco-developed quick-reaction techniques.

This RB-47L project is part of the pioneer efforts in the field of Electromagnetic Reconnaissance, Mobile Range Support, Airborne Command Communications and Control being done by Temco Aerospace Systems, P. O. Box 1056, Greenville, Texas.

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MEMORY PLANE, left is one of 39 utilized in random-access, non-destructive advance memory for Gemini computer. Plane consists of 6,000 multi-switched electronic switches. Switch is similar to a classic transformer, in that it is composed of square loop magnetic material enclosing two spacers. Memory array, right, is thick of 39 memory planes making up RANRAM system for Gemini computer. Array is shown prior to interconnection of individual planes for addressing and sensing.

Gemini Rendezvous

For the rendezvous mission, the Agena D will be launched into a circular orbit of 150 nautical miles.

Initially the Gemini spacecraft would have an elliptical orbit with an apogee of 318 nautical miles and a perigee of 80 nautical miles.

Rendezvous maneuvering in the sky such as such would require the perigee to 315 nautical miles making it an almost circular orbit.

Naval rendezvous between the spacecraft will require that their orbits be replaced within a dozen feet. But all corrections involving the use of the sidereal rate of the spacecraft will allow the spacecraft to approach the Agena D at an angle of 30 degrees. The Agena D will approach the Gemini spacecraft at an angle of 30 degrees. The Agena D will approach the Gemini spacecraft at an angle of 30 degrees. The Agena D will approach the Gemini spacecraft at an angle of 30 degrees.

Gemini system flexibility is provided in the rendezvous mode by alternate guidance systems. Rendezvous can be accomplished by either nonautonomous guidance using ground tracking system techniques or orbital mechanical guidance using the inertial guidance system.

Capability of both systems will be evaluated individually for late mission or early abort and the other a backup system.

able thrusters which remain in space-coast coordinates.

Initial guidance system unit construction are Advanced Technology Laboratory, Houston, Texas. Long Ranger, attitude display group, Westinghouse, rendezvous radar (Honeywell) inertial navigation unit, and Engstrom Magnetics power supply.

Honeywell and IBM share equal performance responsibility for the initial guidance system.

Chief difficulty is the computer design, since weight and power requirements were not, was incorporating standard electronic advances in virtually all construction techniques into the production prototype, IBM says.

Saturn 5 Test Stand Nearing Completion

Construction of the Saturn 5 booster test stand is nearing completion at Marshall Space Flight Center, Huntsville, Ala. The Saturn booster consists of five Rocketdyne F-1 engines designed to generate 7.5 million lb. thrust.

Completion of the test stand, center of the new S-1B engine test area is scheduled for next year. First booster engine test, probably with only one engine, is scheduled for late January or early February of 1964.

The test stand, which is 185 ft. high, consists of three sections:

- First section is four 144 ft. high ball low concrete legs. Office, shops and instrumentation rooms will be built within the legs. The ballow legs, built with 4 ft. thick, are 47 ft. square at the base and 10 ft. square at the top. Foundation walls for the legs are set into bed rock, 40 ft. below the ground.
- Second section, a steel superstructure, will extend 127 ft. above the concrete legs to a height of 265 ft.
- A 380-ton steel leg derrick, with a 135 ft. long boom, which will be mounted atop the steel superstructure.

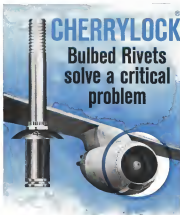
The steel test stand will hold the S-1C stage in a fixed level platform 120 ft. up from the base of the tower. Four steel girders located on the platform will support booster structure. The stand has a steel cradle at its top to 32 million lb. thrust. Booster 150 ft. long with 416 in. dia. can be tested on it.

A water flow of 338,000 gal. per min. will be required to cool the 1,700-ton deflector which directs the engine exhaust gases away from the stand. The deflector has 17,271 holes in its upper side to cool the high pressure water with the coolant phase.

Booster measuring devices will transmit information on 1,200 channels to recording instruments in a Modhouse near the test stand.

General contractor for the Saturn test stand is E. J. Hollen & Co., San Francisco, Calif.

Three firms also are doing the electrical contracting.



CHERRYLOCK®

Bulbed Rivets solve a critical problem

Accepted General Corp.'s Structural Materials Div. will build and deposit two glass fiber-reinforced plastic, solid-propellant rocket motor cases, each measuring 260 in. in diameter and 94 ft long. Case wall will be 14 in. thick. The contract, awarded by the Air Force's Aerospace Systems Div., is for approximately \$8 million.

The two cases ordered to be the largest filament-wound ones ever built for solid-rocket motors, are to be completed in late 1984. Design and materials for the cases will be studied by welding two subscale chambers, 56 in. in diameter and 14 ft long.

Two tubular units will be welded horizontally on a segmental turntable around a wall on a slanting rubber band needed for hydrotesting. The cases will be tested on the welding machine by means of a head placed over them after welding.

The first of the two tubular cases will be tested to destruction. The second will be prepared for shipment to a propellant loading site, indicating that the weld may be built and field cases built in the future.

The two 260-in.-dia. chambers and associated hardware will be built and tested under Air Force supervision at Ford St. Johns Corp., San Pedro, Calif., which offers access to water transportation.

Only Bulbed Cherrylock Rivets were able to solve a critical problem of sonic vibration for one aircraft manufacturer.

Solid aluminum rivets failed in the original construction. Several blind rivets were field tested in flight, but only the Bulbed Cherrylock was satisfactory on this important repair job.

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For technical information on the Bulbed Cherrylock Rivets, write Cherry Rivet Division, Townsend Company, Box 2827N, Santa Ana, California.

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Solid-Rocket Cases

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NASA Funds Study Of MSC Local Impact

Detailed study to determine the physical effects National Aeronautics and Space Administration's Marshall Spacecraft Center will have on the Houston area when the new Clear Lake area becomes operational is the subject of a study being conducted by Southwest Research Institute under a NASA contract.

The study will include three phases that MSC will go through before planning to create good relations with the communities in the Houston area and to determine that its program will not be adversely affected by local environmental conditions.

The study aims to:

- Determine the effects of the current environmental conditions on and adjacent to the site.
- Predict the future developments on the site and in the surrounding area and the effects of these developments.
- Recommend an environmental program of monitoring, sampling and testing. These are required such as to background radiation from radioactive tracer materials, which may be used in MSC testing, and environmental patterns of water and air pollution.

GD/A Satellite to Study Space Phenomena

By Irving Stone

Los Angeles—Satellite for investigating space-environment phenomena is prepared for launch into a 500-mile orbit by the Chrysler Aerospace/Astronautics firm from Cape Canaveral in December as a payload on one of the Atlas.

Orbital altitude will be attained with an additional boost from a solid-propellant, subpropellant rocket motor carried by the GD/A satellite.

Satellite will be the first to be fitted with solar cell panels configured in hemispherical domes at the ends of a cylindrical center section of the payload.

Designated SATAR for multiple air-spectrum research, the payload will contain about 250 lbs. of instrumentation, in addition to the solar panels and associated power systems equipment. Instrumentation will include magnetometers, radiation detectors, hemispherical measurement equipment, nuclear radiation detectors, radiometers, and non-ionizing detectors.

SATAR is sponsored by the Air Force Office of Aerospace Research at an addition to the existing Atlas contract. GD/A is contributing to the cost of the orbital satellite, and there is a possibility of a substantial number of follow-on satellites in the program.

SATAR will measure 44 in. in overall length. The cylindrical center section measures about 27 in. in dia., and the hemispherical solar panel domes extend 12 in. beyond the center section. The solar cell configurations will eliminate the need for latching and packaging a flat solar cell panel configuration for storage during launch. It also will eliminate the need for attitude control in the point the solar cell panels toward the sun. SATAR will transmit commands in orbit and receive data down on its longitudinal axis.

Solar cell substation will be the most challenging requirement of the satellite configuration. Each solar panel will be 12 ft in length and 12 ft in width after mounting on a 12-ft aluminum alloy substrate approximately 65 mils thick.

Solar cells will be blue-N-type (negative, or positive semiconductor) silicon with resistance of 1 to 2 ohms. Cells will be expected to have a conversion efficiency about 10.5% AMO (air mass zero), including cover glass. This is considered a relatively high efficiency compared and indicates that about 15% efficiency must be attained before adding optical filter to the cells.

Optical glass applied to individual solar cells will cover at least 95% of the cell active area and be 40 mils thick. The blue-reducing coating will

point to the solar cell faces will reflect the undesirable ultraviolet portion of the solar spectrum. This filter coating will have an average transmittance of less than 1% for wavelengths between 0.300 and 0.400 microns. For a wavelength of 0.410 ± 0.015 the transmittance will be 50% and for wavelengths between 0.500 and 1.000 microns will be a minimum of 90%.

Average distance to solar radiation is 0.2 to 2.5 astronomical units will not exceed 0.52. Average distance of the solar cell tray surface, with its glass cover, will not be less than 0.52 in the temperature range of 0 to 100°C.

Solar panels will incorporate reflecting shades to prevent the cells from becoming a load to the remaining portion of the power generation system.

Cell arrangement on each panel will consist of 48 strings of 60 cells connected in series, but each group of 48 strings will be connected in parallel. Each group will use at least one isolation diode.

Conditions for testing the panels will include a brightness light source at a color temperature of 2,500K, an illuminance

of 100 mcd/m² at 100 m, and a panel temperature of 25°C. With light rays parallel to the longitudinal axis of the solar cell dome, the panel is required to deliver not less than 21 w at 25 v and 888 milliwatts. With the light rays normal to the longitudinal axis the panel will have to deliver not less than 17 w at 25 v, and 888 milliwatts.

In the 100 mcd/m² test, with an aspect ratio of 20:1, maximum initial power delivered will have to be not less than 24.6 w at 25.8 v, and 910 milliwatts, when temperature is stabilized at 80°C. Aspect ratio is the ratio of the solar panel area to a specified angle across the equivalent amount of panel required to give the stated output when the light is normal to the panel. Under these same conditions, with 10% reduction in power after three months in orbit, minimum power delivered will have to be not less than 22 w, at 24 v, and 920 milliwatts.

In orbit, the solar panel will be required to operate for 16,000 h while stabilized at a temperature within the range of 40° to 160°.



CONFIGURATION OF SATAR satellite for aerospace research is shown in detail. SATAR is prepared to be launched by payload on a General Dynamics/Astronautics Atlas booster and injected into 500-mile orbit by additional boost from the satellite's solid propellant, subpropellant motor. The satellite's cylindrical center section will be capped with hemispherical solar panels. SATAR will transmit telemetry in its orbit path. The satellite is sponsored by the Air Force Office of Aerospace Research.

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AERONAUTICAL ENGINEERING



FREQUET 941 STOL TRANSPORT is shown taking off on a hot flight (left). Note extension of full span, double-dotted line. Folding span (right) is made with flap fully extended (no deflected flap) after. Maximum deflection of main flap is 30° up.

Foreign Interest Revived in Breguet 941

By Warren C. Welles

Paris—Foreign and domestic interest in the Breguet 941 deflected airplane STOL transport has been revived by the French government's recent \$25 million supplementary budget request to begin production of the aircraft (AW June 24, p. 31).

Negotiations are under way with governments and private companies, and Breguet expects to announce the first orders for the aircraft before the end of 1965, according to Paul Gailford, Breguet's commercial director.

Target date for starting production of the 941 is still undecided, Gailford said. So is the status of a proposed cargo/passenger version, designated the 942.

Breguet is discussing with French civil aviation officials the building of several 941 factories, which, along with associated landing gear and wing mounting structures, is the principal difference from the 942.

Production rate of the 941 has been tentatively set at two aircraft per month. Gailford attributes the return figure to the company's desire for long-term production. This rate could be increased to a maximum of 4 aircraft per month if the demand is sufficient.

Current price quotation for other versions of the aircraft is \$1,951,600. Cost of the 942 is projected five less, is offset by that of the 941's new door and wing.

These figures represent a substantial increase over the previously quoted price of \$800,000 (AW July 12, 1963, p. 103), an increase, Gailford said, due

to growing costs of labor and materials. Compared with other aircraft, however, the latest price is only about 15% higher than conventional non-STOL airplanes in the same payload class, Gailford said. Total expenditure on the program of \$16 million is 50% funded by French government and 50% Breguet and associated contractors.

Production version of the 941 and 942 will be powered by four Turbomeca Turmo 303 five-laborer turboprop engines rated at 1,480 shp each, a gain of about 210 shp per engine over the 941 prototype's Turmo 310s. The new engine has been limited by Turbomeca and uses the same governor as the Turmo 303 engine in the Sud Super Pécure heavy helicopter.

Mean time between overhaul (MTBO) is listed at 1,500 hr, and growth potential is estimated at over 100 shp.

The two-stage free turbine is driven at 19,000 rpm. It, in turn, delivers power to a propeller shaft located above the turbine case through a reduction gear—which lowers the rotation speed to 6,000 rpm—and a transmission containing of a two-wheel and a dog clutch unit.

A constant propeller selection gear located just behind the propeller hub further reduces the shaft speed to the nominal propeller rate of 1,200 rpm.

Propeller interconnection is a safety mechanism designed to prevent thrust asymmetry, at the event of engine failure—achieved with a flexible steel cross shaft developed by Hispano-Suiza. The shaft extends the entire length of the landing gear between the two outboard engines.

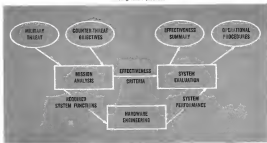
The shaft is divided into four sections which are joined at the four gear boxes. Included the two outboard engines and clamped at a transmission box in the fuselage, which also contains the propeller pitch governor constant speed unit. The two outer shafts are each 14.1 ft long, while the inner shafts—which bear the brunt of loading due to wing deflected slung under carrying load—each are 12.1 ft long. The cross shaft passes through 15 shaft of bearings and the wing landing edges are removable for maintenance, repair.

In the event one engine fails, the propeller is governed by the remaining three engines through the cross shaft with a direct loss of about 17%. This is particularly important during STOL operations where asymmetric thrust

Br. 941 Specifications	
Wingspan	76.7 ft
Length	71.0 ft
Height	30.7 ft
Wing area	497 sq ft
Mean aerodynamic chord	12.19 ft
Mean area	6.75
Empty weight	25,920 lb
Max takeoff weight	45,000 lb
Max landing weight	45,000 lb
Max fuel and oil	17,900 lb
Max speed	340 kt
Cruise speed	275 kt
Max altitude	30,000 ft
Normal altitude	10,000 ft
Cargo hold	
Length	35.6 ft
Width	5.1 ft
Height	9.4 ft
Volume	2,330 cu ft

weapon system operations analysis

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LEADING EDGE OF 941 wing is reasonable for cross-shaft maintenance. Means that between overhead for shaft system is about 1.580 in., according to company. Breguet hopes to increase this to 1.800 in.

could have disastrous consequences. Propeller revision is also presented in this manner.

Conversely, one of the propellers can be feathered and its engine will continue to drive the three remaining propellers, entailing a 30% loss in thrust. Power loss in the cross shaft system is less than 10%.

The current two fixed-bladed variable pitch Breguet-Rohrbach propellers with diameters of 14.76 ft. These give 44 lb of thrust per horsepower and a cruise efficiency of 80%. The propeller system consists in three modes: • Constant speed mode, employed during conventional cruise operation. The propeller pitch governor continuously adjusts constant propeller pitch to maintain the desired rpm. The cross shaft also ensures absolute synchronization of all four propellers.

• Differential pitch mode supplementing control outflow during STOL flight. Pitch of the two outboard propellers differentially changed by movement of the pilot's stick. A left displacement increases the pitch-and decreases the thrust-of the outboard starboard propeller and correspondingly decreases that of the port outboard propeller, and vice-versa for a movement of the stick to the right. This action produces yaw and roll due to increased fan thrust on the outside of the turn and propeller slipstream deflection, respectively.

• Transgressive mode permits approach slopes as high as 11.5 deg. during precision landings. Outboard propeller pitch is conventionally offset with respect to inboard propellers, reducing the blown slipstream on the outer wings to zero and boosting the inner-wing slipstream. This causes an increase in induced drag and thus steepens the flight path.

• Reversal mode is activated manually immediately after touchdown. It involves no change in the throttle setting and thus no drag.

The propeller pitch control system is basically a complex mechanical servo mechanism employing mechanical feed back. Inputs consisting of cross shaft speed—depending on the mode of operation—propeller thrust signal, differential pitch signal from the pilot's stick or a pilot signal during approach are used mechanically by differentiated gears on the pitch governor. Four mechanical output—on for each propeller—are mounted to the proper pitch actuator by springs, setting shaft linkages and consist of similar displacements at these four shafts. Displacements are downward in the differential pitch and approach (transgressive) modes, and reduced in the constant speed and reversal modes. Rotating shaft linkages are connected to the hydraulic pitch actuating mechanism by means of screw pins.

Feedline of the 941 is of constant

torsionless mid-length construction and has a rectangular cross section.

Wing structure consists of three stiffened web spars covered on top and bottom by honeycomb sandwich panels in four section bays. Aerial designations is NACA 63A 410, distributed locally by the constant leading edge integral fuel tanks situated within the fuselage have a maximum capacity of 1,662 U.S. gal. Leading edge slots are not used since Breguet found the center to be equally effective for high lift.

Thrust of Breguet's deflected slipstream approach in STOL lies in the large double deflected flow wings which extend the entire length of the wing leading edge. Maximum deflection of the outer flaps is 100 deg, and the outer flaps which are also used as airbrakes can be extended to 70 deg.

This technique does not involve any



TWO LIGHT FORMER ARMY trucks are driven into B-57 cargo field. Aircraft's 15,000-lb payload capacity permits carrying an armed car. Note airbrake left on backwing.



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problems with slowdown during engine run-up or ground run. Request slowness. No ground engine or without acceleration has been observed.

Additional roll control is derived from the differential-actuated spoilers located on the upper surface of the wing forward of the outer flap. Transverse structure of the empennage is similar in construction to that of the wings. Control surfaces consist of a double rudder and conventional ailerons. In addition, the forward stabilizer is hydraulically adjustable in three preset positions for takeoff, cruise, and landing and it has an in-cabin-controlled locking edge to prevent flutter vibration during the STOL operation.

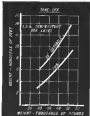
Flaps and all control surfaces are power-actuated by the 3,000 psi hydraulic system—which features two independent circuits—and operates all the major lateral functions. Also installed in the system are artificial feel devices fitted with force-trimming outputs on all three axes.

The Meteor built landing gear is specially designed for rough-field operation and consists of a dual-wheel nose gear and tandem main gear.

Wheel base is 216 ft and track is 11.5 ft.

Main gear wheels are mounted on a shock-reversing strut connected to the new horizontal linkage system, which functions as both a shock absorber and a reaction mechanism. Maximum stroke of the main gear is 25 in., and least vertical velocity is 12.12 ft/sec. Reaction is instantaneous during flight; the wheels provide vertical motion before their travel ceases.

Individual struts for both nose and main gear also function as jacks, allowing slowness of the aircraft ground attitude to facilitate engine starting.



BREGUET 941 TAKEOFF capabilities are shown in the above chart.



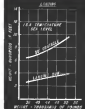
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LANDING CHARACTERISTICS of the Breguet 941 are shown depicted in chart form.

Breguet says the most extreme pitching moments in the STOL envelope have been translated into lift. The aircraft remains stable with a center of gravity displacement of 25.5% of the mean aerodynamic chord. Laterally, the Breguet roll mode is well-damped. Chances of stall in ground and in approach by increasing banking. In the STOL configuration with the gear down, stall occurs at around 36 ft. Breguet says.

Operation of the B-41 has been divided into three general missions.

• Assault. The assault can deploy 55 fully armed troops or an equivalent 6,600 lb payload into an unopposed field close to a combat area. Mission profile calls for a 750 knot to reduce without midcourse tracking. First 161 mi of the outboard leg is flown at 10,000 ft and the remaining distance at 1,000 ft. Cruise speed at both altitudes is 215 kt. Return path would follow the high low pattern in reverse. Maximum initial takeoff weight is 44,000 lb, and maximum landing weight—meaning that the aircraft returns with a 6,600 lb payload—is 37,000 lb. Midpoint landing over a 10-ft obstacle requires a distance of 700 ft and weighed itself a distance of 570 ft in about 35 ft obstacle.

• Logistics support. For this mission, the B-41 operates point-to-point over a 900 mi or other length with 11,600 lb payload at 18,000 ft altitude, and at the normal cruise speed of 215 kt. Takeoff gross weight is 52,000 lb, resulting in a takeoff distance of 1,385 ft to clear a 35 ft obstacle. Landing weight is 46,000 lb, requiring 790 ft with a 90 ft obstacle.

• Long range. In missions that can be flown from proposed fields, the aircraft will be able to land 16,000 lb, cruise altitude and a 215 kt cruise



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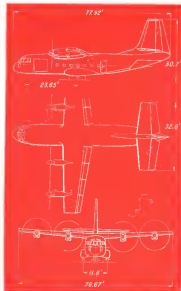
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THREE VIEW SHOWS dimensions and configuration of Fagot 945 STOL, transport

speed. Takeoff gross weight is increased to 55,400 lb and takeoff distance over 135 ft absolute to 1,710 ft.

Zero-payload turn range is 1,680 feet/sec, which can be boosted to 2,700 feet/sec by installing temporary additional linkage in the cargo hold.

Side flight of the B-941 in STOL configuration is very much like that of flying in a helicopter. During a recent demonstration at the French Air Force's Villacoublay Field near Paris, the B-941 pilot put the aircraft through a series of tight, well-coordinated turns at approximately 64 ft/sec with no apparent instability or loading. Numerous testing for slow flight in the STOL

configuration is more than 10,000 ft. Takeoffs and landings were completely devoid of the climbout and approach slopes. Touchdowns were gentle and deceleration of 1g during propeller reversal did not prove bothersome.

Takeoffs and landings were made diagonally across the runway. The aircraft's STOL capability is such that the pilot did not even align the aircraft with the center strip. The landing and offloading of passengers, the two port propellers were disengaged while the engines were left running, a feature which should permit rapid turnaround for the aircraft.

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more than 500 takeoffs and landings have been logged by the 941 prototype to date. The company also has devoted considerable time to study of STOL landing techniques and touchdown point dispersion. In the latter, touchdown point was found to vary approximately ± 50 ft in a statistical angle of 50 landings performed by different pilots.

Landing approach slope selection is aided by a pictorial indicator on the cockpit. This device consists of three vertically arranged lights connected to a differential potentiometer angle of attack sensor located on the landing gear. Since the approach is made at an approach constant angle of attack, the pilot can determine whether he is above, at, or below the proper slope by noting which of the three lights is on.

Another landing aid which proved useful was a terrain sensor similar in principle to that used on aircraft carriers.

The technique reduced landing point dispersion to less than ± 50 ft.

Only difficult part of pilot training for the B-49 is likely to be the achievement of sufficient accuracy in landing plus the necessity for a considerably qualified pilot to become accustomed to landing at 67% of zero-zero point.

Bergert feels that 25 landings should be adequate for conversion to the aircraft.

AFOSS Awards

American Research Institute, Santa Fe, N.M., has been awarded the American Society of Aeronautics and Astronautics Award for the study of non-destructive and active of aerial failure to reactivated Research Institute of Aeronautics, Santa Fe, N.M., for the study of non-destructive and active of aerial failure.

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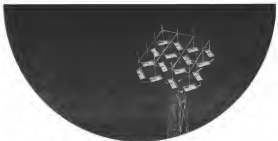
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The page opposite describes one of the many Dalmo Victor achievements. Dalmo Victor and employees of unusual ability are needed in further air and other Dalmo Victor companies. If you would like to work in this exciting atmosphere, send away for the many advantages of living in the San Francisco Peninsula area. Investigate a career with Dalmo Victor. It can be most rewarding.

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DALMO VICTOR PUTS "SATAN" TO WORK FOR NASA Project "Satan" improves capability and reliability for data acquisition, telemetry and command. Thus NASA's Goddard Space Flight Center's worldwide Satellite Tracking and Data Acquisition stations perform pre-designed tasks on the various satellite programs—automatically and with minimum maintenance. □ Project "Satan" is an example of the key role played by Dalmo Victor in development work with NASA scientists, and in the manufacture of high-performance aerospace systems. DV is in the vanguard of new developments in its major product areas. If you are interested in becoming part of these challenging programs, Dalmo Victor currently is inviting applications from qualified scientists and engineers. For further information contact, Director, Scientific and Engineering Personnel.

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THEORETICAL CALCULATIONS TAKE YOU ONLY SO FAR: Comes a time when paperwork gets the air in high reality and tunnel tests. In those tests, air at Mach 3 plus speeds goes against the stark realities of serious wing/body configurations. Result: a formula is converted into form, mathematics into curved metal.

At Lockheed California Company, a team of engineers and scientists currently is engaged in an intensive study of flight and operational characteristics involving America's first Super sonic Commercial Air Transport (SCAT). These studies include evaluation of flight and wind tunnel data to determine performance criteria and mission capabilities of tomorrow's airliner.

Planned to operate on the wings of space, SCAT presents many unique problems. The solutions are doubly important, since they involve factors of practical application to airline operation. Advanced conceptual studies are also being carried on with V-157G, aircraft, hypersonic fighter aircraft and rotorcraft helicopters.

Projects such as these at Lockheed California add up to an exciting future—make it yours!

PAPERWORK GETS THE AIR

QUALIFIED SCIENTISTS AND ENGINEERS are invited to investigate immediate openings in: Aerodynamics, Thermodynamics, Dynamics, Astrodynamics, Structural Dynamics, Propulsion, Operations Research, Weapons systems evaluation, cost effectiveness, Human Factors, Reliability, Spacecraft, Population (Missiles and Space Vehicles), Controls and Guidance.

Write: Mr. E. W. De Launay, Manager, Professional Placement Staff, Dept. 1107, 2400 N. Hollywood Way, Burbank, California. An equal opportunity employer.

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CALIFORNIA COMPANY

Specialty • Advanced Aircraft • A.S.T.
A Division of Lockheed Martin Corporation

U.S. Business & Utility Plane Shipments

(March 1963)

Office & Model	No. of Units	Net Weight
Aero Commander 440B & 440P	4	50,000 lbs.
Boeing Stearman	2	10,000 lbs.
Cessna 172	10	10,000 lbs.
Cessna 180	10	10,000 lbs.
Cessna 182	10	10,000 lbs.
Cessna 185	10	10,000 lbs.
Cessna 188	10	10,000 lbs.
Cessna 190	10	10,000 lbs.
Cessna 195	10	10,000 lbs.
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Cessna 215	10	10,000 lbs.
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They started with a concept. It's been refined to say the least. Hestel started with a concept, too—the honeycomb configuration (highest strength-to-weight ratio known and other excellent properties). We have also seen some refinement.

SOME HERCEL HONEYCOMB APPLICATIONS—1. Abrasive Heat Shield Matrix (heat resistance and strength) 2. Cassia Energy Absorption (boats, equipment, alternative fuels) 3. Common Bulkheads (structural strength, insulation) 4. Large Sandwich Cylindrical Structures (at least 25% more efficient than monocoque) 5. Stabilizing Pans (non-pro deflected) 6. Propellant Tanks (honeycomb for structure) 7. Nuclear Vehicle (damaging, strength)

WHAT DOES THIS MEAN TO YOU? Of all honeycomb manufacturers **only** Hexcel offers:

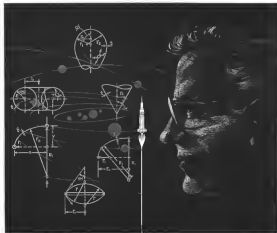
- Largest R&D staff for honeycomb technology
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If you hold a degree and have experience in any of the disciplines listed on the page opposite, and if you think in terms of tomorrow as well as today, you will find a congenial, professional climate at Chrysler where you are encouraged to translate your creative ideas into action.

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Send your resume, in complete confidence, to Section A-5, Personnel Department, at the location of your choice:

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SPACE DIVISION



CHRYSLER CORPORATION

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AERODYNAMICS

Aerodynamic studies on Cometary-like bodies, stability, stability and so forth, for re-entry vehicles, thermal protection, shock waves and expansion, design, zeroth order, second order, third order, etc. (spacecraft, planetary, and atmospheric re-entry, etc.) (spacecraft, planetary, and atmospheric re-entry, etc.) (spacecraft, planetary, and atmospheric re-entry, etc.)

HYDRAUTICS AND PNEUMATICS

Major studies and design studies, research, development, program, design, etc.

ELECTRICAL AND ELECTRONICS

Design, design and development, research, development, program, design, etc. (spacecraft, planetary, and atmospheric re-entry, etc.) (spacecraft, planetary, and atmospheric re-entry, etc.) (spacecraft, planetary, and atmospheric re-entry, etc.)

LAUNCH FACILITIES

Design, design and development, research, development, program, design, etc. (spacecraft, planetary, and atmospheric re-entry, etc.) (spacecraft, planetary, and atmospheric re-entry, etc.) (spacecraft, planetary, and atmospheric re-entry, etc.)

MANUFACTURING

Design, design and development, research, development, program, design, etc. (spacecraft, planetary, and atmospheric re-entry, etc.) (spacecraft, planetary, and atmospheric re-entry, etc.) (spacecraft, planetary, and atmospheric re-entry, etc.)

PROGRAMMING

Design, design and development, research, development, program, design, etc. (spacecraft, planetary, and atmospheric re-entry, etc.) (spacecraft, planetary, and atmospheric re-entry, etc.) (spacecraft, planetary, and atmospheric re-entry, etc.)

QUALITY CONTROL

Design, design and development, research, development, program, design, etc. (spacecraft, planetary, and atmospheric re-entry, etc.) (spacecraft, planetary, and atmospheric re-entry, etc.) (spacecraft, planetary, and atmospheric re-entry, etc.)

RELIABILITY

Design, design and development, research, development, program, design, etc. (spacecraft, planetary, and atmospheric re-entry, etc.) (spacecraft, planetary, and atmospheric re-entry, etc.) (spacecraft, planetary, and atmospheric re-entry, etc.)

SYSTEMS ENGINEERING

Design, design and development, research, development, program, design, etc. (spacecraft, planetary, and atmospheric re-entry, etc.) (spacecraft, planetary, and atmospheric re-entry, etc.) (spacecraft, planetary, and atmospheric re-entry, etc.)

SYSTEMS INTEGRATION

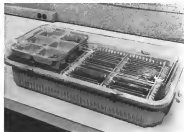
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TESTING

Design, design and development, research, development, program, design, etc. (spacecraft, planetary, and atmospheric re-entry, etc.) (spacecraft, planetary, and atmospheric re-entry, etc.) (spacecraft, planetary, and atmospheric re-entry, etc.)

TRAINING

Design, design and development, research, development, program, design, etc. (spacecraft, planetary, and atmospheric re-entry, etc.) (spacecraft, planetary, and atmospheric re-entry, etc.) (spacecraft, planetary, and atmospheric re-entry, etc.)



Container Holds Circuit Boards

Don't fix containers hold various components and printed circuit boards during assembly in assembly operations (below). Designed by Vector Co., the containers come in two sizes with removable plastic partitions to accommodate a variety of items. Locks hold the transparent cover in place to keep it closed. Even for small components are designed to hang from fixtures on assembly bench. Containers are made 11 in. wide by 6 1/2 in. high by 11 in. 14 in. length.



guided as entirely or in part unguided. Whether ideas were not given a chance to receive user test results and offer what might be solid concepts or not, before the reports were distributed by IDEP.

Now, about 90% of IDEP participants are encouraged to submit reports of their test results in the development of the initial component, before they are sent to IDEP, and for secure but for comment. The results are sent to them, weeks to submit the comments on the report and then normally are appended to the next test report.

Below the more circulated report can present both sides of any dispute raised on a parts test. At times, prompt response from the vendor has caused withdrawal of a report for retesting of parts.

About two years ago (AV Nov. 6, 1961, p. 78), IDEP began a program aimed at doing away with defective in planning, test, test, the modified per IDEP or IDEP program. Under IDEP's methods, submitted plans of contemplated tests to do to give other contractors an opportunity to supply results that can depicting clearly. Before records IDEP has 100



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AC has assembled one of the finest technical teams in the country. They are growing rapidly in such outstanding projects as APOLLO, Minuteman Guidance, TITAN II & TITAN III Inertial Guidance and the B-57C/D Bombing Navigation System. AC will also provide local and professional education for you. You enjoy the satisfaction of accomplishment at AC . . . not only in the job you do, but the advancement that comes from achievement. As a member of a division of General Motors, you will share outstanding corporate facilities.

Review the opportunities listed below. This content is for an immediate evaluation of your qualifications.

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Development, Manufacturing)

SYSTEMS ANALYTICAL ENGINEERS—Perform analytical studies of inertial guidance systems, including analysis of system performance, time constants, velocity errors and rate of acceleration. Develop time, velocity and position errors on digital and/or analog computers. Conducting trajectory, drift, and pointing guidance equipment. BS or MS in EE, with at least 2 years experience in inertial guidance systems.

SYSTEMS RECONSTRUCTION ENGINEERS—To design and maintain inertial guidance systems in aircraft. BS or MS in EE or ME, with at least 2 years experience in inertial guidance systems. BS or MS in EE or ME, with at least 2 years experience in inertial guidance systems.

ORBIT DESIGN & ANALYSIS ENGINEERS—To design and analyze orbital trajectories. BS or MS in EE or ME, with at least 2 years experience in orbital design and analysis. BS or MS in EE or ME, with at least 2 years experience in orbital design and analysis.

DIGITAL COMPUTER ENGINEERS—Design, develop, and test digital computers for inertial guidance systems. BS or MS in EE or ME, with at least 2 years experience in digital computer design and development.

CYRO ENGINEERS—Design and develop cryogenic systems for inertial guidance systems. BS or MS in EE or ME, with at least 2 years experience in cryogenic systems.

INTEGRAL SPECIALISTS—Design and develop integral systems for inertial guidance systems. BS or MS in EE or ME, with at least 2 years experience in integral systems.

SCIENTIFIC PROGRAMMERS—Develop and test scientific programs for inertial guidance systems. BS or MS in EE or ME, with at least 2 years experience in scientific programming.

SYSTEMS ENGINEERS—To assist in the design of

inertial guidance systems and general systems equipment including the development of new systems. BS or MS in EE or ME, with at least 2 years experience in inertial guidance systems.

MECHANICAL DESIGN & DEVELOPMENT ENGINEERS—To design and develop mechanical systems for inertial guidance systems. BS or MS in EE or ME, with at least 2 years experience in mechanical design and development.

EQUIPMENT DESIGN ENGINEERS—Design and develop equipment for inertial guidance systems. BS or MS in EE or ME, with at least 2 years experience in equipment design and development.

DEVELOPMENT ENGINEERS—Perform engineering development, project planning and equipment design. BS or MS in EE or ME, with at least 2 years experience in development engineering.

WEST COAST—Los Angeles

(Advanced Systems Laboratory,
Research & Development)

SYSTEMS SCIENTIFIC PROGRAMMERS—To assist in the design and development of inertial guidance systems. BS or MS in EE or ME, with at least 2 years experience in systems programming.

SYSTEMS RECONSTRUCTION ENGINEERS—Design of inertial guidance systems. BS or MS in EE or ME, with at least 2 years experience in systems reconstruction.

For further information, write and
request to Mr. E. J. Davis, Director
of Research and Professional Staffing,
AC Spark Plug, 4000 AC Spark Plug
Company, General Motors Corporation,
Warren, MI 48090.

AC SPARK PLUG

The Electronic Staffing Division
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EAST COAST—Boston

(Advanced Systems Laboratory,
Research & Development)

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the individual responsible for scheduling test reports and handling IDGP reports. If the coordinator is a live wire, IDGP usually has been a success in the history.

Purity is a trademark of the acceptance program has gained, IDGP coordinator is becoming a recognized position in government contractors' circles. It is not uncommon, Barke says, for an IDGP coordinator at one company to consult to that position at another company.

IDGP encourages coordination to correct one another and discuss various IDGP problems. Recently, the first of what is planned as a series of annual meetings of all IDGP coordinators was held in Dallas at the initiative of industry representatives.

Their purpose was to organize technical views, to discuss common problems and to supply IDGP data centers with needed feedback on the operation.

A contractor-owned steering group is being organized to handle basic industry concerns to IDGP centers in various locations, with parts suppliers and to partly shoulder necessary tasks not handled in IDGP.

Barke has assembled a team of letters and sketches to back up his claims that IDGP has saved government contractors more millions of dollars. Several specific cases with estimated dollar savings for each, assembled earlier in the program, were given prominence (ENR Nov. 5, 1961, p. 70).

A single source struggle is one of many letters, then one from a large aerospace manufacturer. Among the quantitative and qualitative savings attributed to IDGP is:

- Consolidation of various loads on printed circuit assemblies, based on stress, load and panel orientation, saving materials and glass capacitors and rare cost materials at a total savings of \$15,000.

- Shortening of test time between station and return after a series of available IDGP reports, providing \$4,000 in savings.

- Lower selection of vital components and change in the number of tests in the position to potential material savings.

Besides IDGP contributions to service contractors, data generated by contractors of Naval Air Materiel and Space Air Materiel is being used in the program.

Additionally, IDGP data currently is being used by Auto Electronics Corp., a Sharon, Ohio, Civil Aeronautics Administration service group, in support of a NASA program. Auto Electronics is using IDGP data to establish uniform material specifications, emphasizing of parts to perform an specific aerospace of NASA interest.

Even earlier month IDGP traces to

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Tucson Engineering Laboratory
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AVIATION WEEK & SPACE TECHNOLOGY, July 15, 1962

DEFENSE CONTRACT AWARDS SECOND FISCAL QUARTER SUMMARY—1963

Now available free from AVIATION WEEK & SPACE TECHNOLOGY are summary reports of defense contract dollar awards covering the second fiscal quarter of 1963. These reports show defense dollars awarded in 179 product/system categories as compiled by Frost & Sullivan, Inc. Information is also available on the second fiscal quarter 1962; third fiscal quarter 1962; fourth fiscal quarter 1962; and first fiscal quarter 1963.

Reports are available on an individual request basis in the following system areas:

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preparing contract a listing of IDLP reports, which covered with an order comparison volume, compare an index of all distributed IDLP documents—PREFP reports, current reports, underpin standards and practice reports. This are divided within parts type with various smaller part number, standard part number and data all included. Control information provides details of the part, its principal of operation, material, characteristics.

These appear to be a possibility that a government agency may utilize IDLP data to a tape from outside as an initial step toward setting up a system for automatic machine retrieval of available IDLP information.

IDEP has grown as justified and in cooperation with the Navy's Coated Nuclei Data Exchange Program (CNDEP) sponsored by the Bureau of Weapons. The two work closely, many of their personnel being shared and their data exchanged so that IDEP is joint endeavor those generated by CNDEP instructions.

Partners in IDEP generally are government funded activities, whose tests would be of value to other government sponsored activities. The funding agency notes on the acceptability of each participant.

The Annual Services Technical Information Agency (ASTIA) began publishing late last year as its Technical Abstracts an abbreviated form of IDEP research results in selected tests. The ASTIA reports are not made available as quickly as the original documents.



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Cooper Sightings

In your story of the MAV flight (AW May 17, pp. 24, 25), you reported that astronaut Cooper said that he could see some objects in orbit as he took on a high way turn to escape. I find this difficult to believe.

If we assume an altitude of 330 mi. and a beam on the ground that is 10 ft square, we find that the angle subtended by the beam to us observer at that altitude to be an order of magnitude smaller than the 1 mi. of arc generally taken in the limit of average resolving power. In short, unless Mr. Cooper has visual acuity worthy of a squirrel study in itself or he was making a good case for a missed transverse satellite. In no story of this type have I seen reference to any optical aid on board, so I must apprehensively conclude that those "observations" were hallucinations. Could NASA's medical men close up the confusion for us?

HAROLD DRENNER
Schaffsville, N. Y.

(As Reader Doublet indicates there is considerable skepticism over Mr. Cooper's report of photographing small objects from his capsule (AW June 17, p. 58, July 1, p. 11). Moreover, the official NASA viewpoint is that Cooper was the first U.S. pilot to fly in space on a visual aid, and did so on December 16, 1968, when U.S. Mercury flights for short comparison. Consequently, the agency is withholding payment on whether or not Mr. Cooper was having hallucinations until additional flights are made. Meanwhile both astronauts and biomedical specialists are making separate tests to attempt to develop theory which might explain this report—Ed.)

The episode of Dr. W. R. Adey (AW June 17, p. 14) is the effect that it was "absolutely impossible" for astronaut Cooper "to see 10 ft or the object from an altitude of 330,000 ft." An aside opens to challenge. For example, I—a thoughtful individual—have often been able to distinguish 600 yards from over 100 ft. and millions of quiet children have been heard clearly even 15 ft. Squads at over 200 ft. Park authorities could the possibility of objects at a distance of nearly 1,000,000 times their diameter! The size rightings, a least correct statement that that for the satellites, were made close by or nearby.

In comparison, Cooper's beam—made 10 ft. and best order may be targets at 1 ft. or 33,000.

Only the greater typical resolution and the resolution of 4 or 5 m. of single eye on level air were between the astronaut's eye and the earth. The human eye has significant methods of perceiving objects whose actual angle is smaller than the eye's resolution, and Dr. Adey would do well to rethink his beam comparison, especially considering before further depicting Cooper's report.

PARULI SCHWARTZ
Bridgewater, N.J.

Astronaut Frank confirms the opinion of its readers on the issues raised in the mentioned editorial columns. Addressed to the Editor, Astronaut Frank, 120 W. 42nd St., New York 36, N. Y. To be shown letters under 500 words and give a general identification. We will not print statements directly, but answers of editors will be published on request.

Supersonic Blues

Long the supersonic blues,
Bummer's always making news
"Pretzle is payment," they say,
"To get a man we need Mach Ten."

Don't be fooled by current ads,
Cash ahead the apple cart
C4 super alloys, steel and cast,
Titanium and Douglas?

"Bendless is just the deal,
And then there's always shales still
It may take time, and lots of help,
And at last we have back to it!"

Aluminum is dead and gone
Faster than ever, faster than we want
Who he best, it's just a habit
Not the habit, he the habit!

But now young Jack has set us straight,
With words of wisdom, said at last,
I promise, not the last to such
You give your money, and you're your
Mach?

L. W. MANN
P. O. Box 1012
New Kensington, Pa.

Moon Battle

After editorial gave a name to 340 billion moon program, appearing, Caring Bushchuckers. The big debate is now being up at a Caring Bushchuckers in Moonlighting.

One reason why there is more heat than light coming from this debate is the widespread misleading arguments coming from some naturally prominent agencies. They talk of the moon in the way a mountain climber talks of Mt. Everest. "It is there." They talk of how the 540 billion will have the economy of it some other effort would reach from spending it on schools, medical training for the unregistered, expensive, piece development, helicopter, and VTOL, research, water defense research, or more support for Viet Nam. A few of these men try to show stupid people to Planet X as Christ 9 to the desperate moon.

Heads has not been laid to people whose leaders have led with reality. It is foolish to have space-age wastes in the wall to tell us we are faced of them all. It is dangerous to go to another as Matherly Aid, to be told story of ancient history. We want no future Matherly to write.

Double double, toil and trouble,
Fire burn, and cauldron bubble.

Call it with some before a head
Then the cheap a firm and good.
Try to your General letter
Send The Appropriate letter
Now we know not brother's hand!
Now we see our face all squashed!

There must be some way practical
and second must be spending all the money
in the particular way. Let's have them!
—Caring Bushchucker

W. B. BAKER
5410 W. 10th St.
Los Angeles 93, Calif.

TFX Coverage

Your Letter, page of June 3 indicates that some readers actually support the TFX, not all investigators and your coverage thereof, our own side the accounting, its note of the amount of money to be used through commodity in aircraft, etc., is able.

My vote is one of thanks for the full measure of reporting on their matter. It appears to me to be useful reporting, not in rapidly making them independent newspaper a similar (though less detailed) picture of atmospheric civilian federal of investigators is possible.

It is amusing that the federal executive branch of government can so fully support your supporters.
So it is obviously fortunate if your side heads find to support Senator McGowan, Congressman Voino and their colleagues who logically are going to ensure that the balancing point of the federal legislative branch of government remain a power as the Congressmen agreed.

SEYMOUR H. BARNETT
P.O. Box 315
Pacific Palisades, Calif.

FJ-4B Error

As stated under a one of the front public column in the industry I have found its studies, editorial and general coverage of the news almost without question.
However, I must beg to differ with the opinion on the paper, p. 30 of the June 17, 1968, issue.

Mr. Montgomery shows a Shrike as to surface missile suggested from a pilot as under the wing of a Navy aircraft. Your caption reads this as a Douglas A-1H Shrike, which, of course, is not correct. Close examination of the leading edge of the wing, wing fence, vortex generator around the tail pipe and the laminar flow indicator configuration clearly shows the aircraft to be an FJ-4B.

Having spent a considerable number of pleasant hours at the AED before leaving the server I find it difficult even now to make the two straight!

BARRY W. DRENNER
1030 N. 10th St.
Seattle, Conn.

(Reader Doublet is right—Ed.)

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recorder
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silent light!



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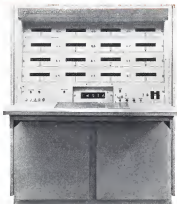
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Writing Speed . . . Greater than 30,000 in./sec.
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designing
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Series 500 Multiparameter Semiconductor Tester. Rapid programming, automatic direct digital readout. Pulse testing. Data logging optional. Can be run by relatively unskilled personnel.

Series 3500 (not pictured). The CAPACITRAN for high speed, automatic test measurements of capacitors by unskilled personnel. Direct digital readout and/or high-low limit programmable indication. High accuracy.



Model 20 Low Current Leakage Tester for semiconductors. Fast, easy go/no-go operation with pass-reject lights or absolute readout. Noise shielded for high accuracy.

Series 900 Card Programmed Semiconductor Data Logger (not pictured). Up to 24 tests in any order. Self-checking. Visual digital readout. Test results available on punched cards, punched tape or automatic typewriter.



Series 250 Go/no-go tester for semiconductors. Allows unskilled worker to test up to 900 units per hour—16 variable go/no-go tests per unit. Pulse testing. Classification chassis for priority sorting available.



Model 50 Pulsed direct reading Beta Tester. Three digit visual display. Two seconds maximum per unit with 2% accuracy. For economical 100% incoming inspection or engineering evaluation. Pulse testing.

Series 100 Diode Sorter and

Series 200 Transistor Sorter (not pictured). High speed multiparameter testing for manufacturing and incoming inspection. Punched card programming. Automatic priority sorting. Handles 3600 diodes or 1500 transistors per hour.

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